

SRS 2H Evaporator Transport Modeling

Valmor de Almeida

Separations and Materials Processing Research Group

Chemical Technology Division

Oak Ridge National Laboratory

dealmeidav@ornl.gov

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RELEVANCE

- Provide insight on the temperature distribution and flow in the evaporator
- Integrate results of concurrent studies on formation of aluminosilicate deposits
- Provide a framework and reference for more advanced transport modeling

GOALS

- Construct and analyze a tractable model of flow and temperature by the end of February 2001
- Produce technical report on 22 March 2001

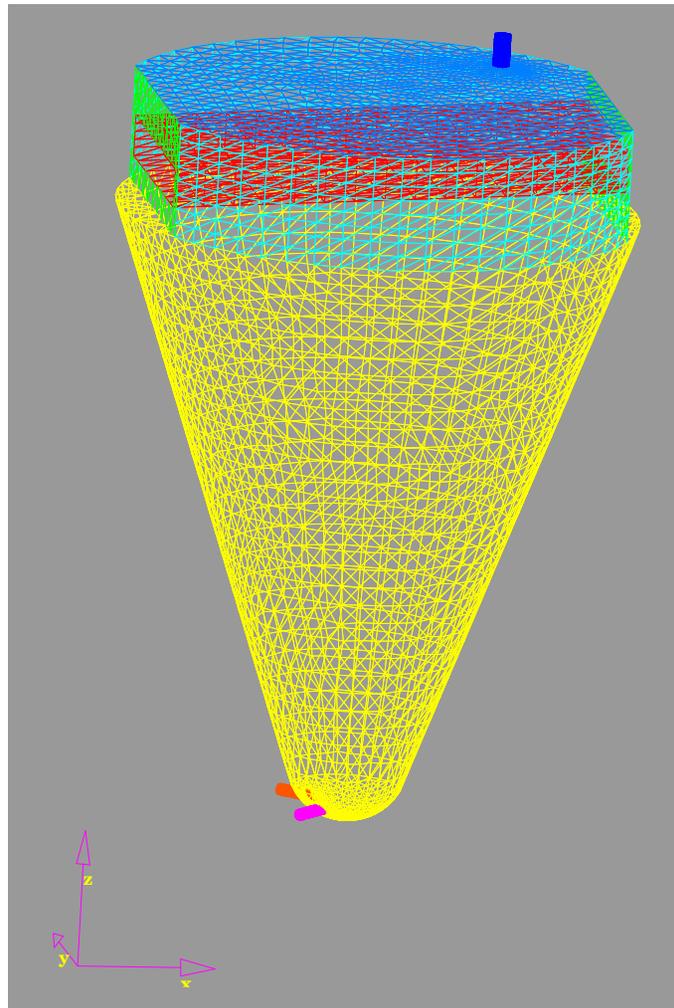
OUTLINE

- Geometric model
- Equations of transport & parameters (dimensionless groups)
- Boundary conditions
- Parametric studies and results
- Solution method & algorithms
- Outlook

GEOMETRIC MODEL

- Three-dimensional
- Based on actual dimensions (tech. drawings & system design description doc.)
- Components included
 1. Artificial shear-free liquid interface
 2. Feed tube
 3. Tube bundle envelope
 4. Cylindric-conical vessel
 5. Discharge lift
 6. Lower lance
- Major items neglected
 - Entire overhead system
 - Warming coil and auxiliaries
 - All other pipes/tubes
 - Internal stay rods and structural beams

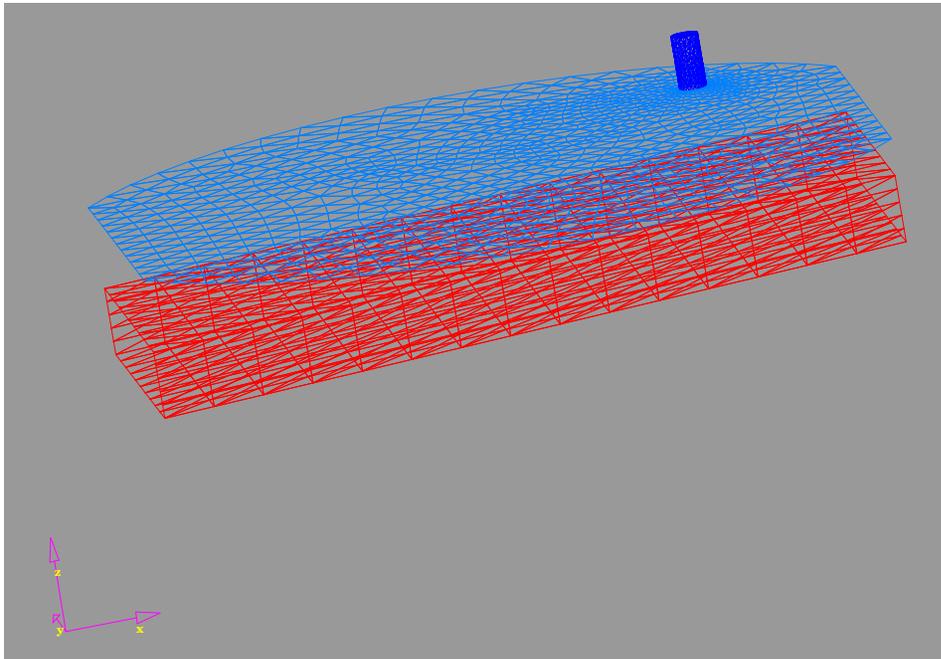
GEOMETRIC MODEL



EAST-WEST VERTICAL CROSS SECTION (DWG No. D-10578)

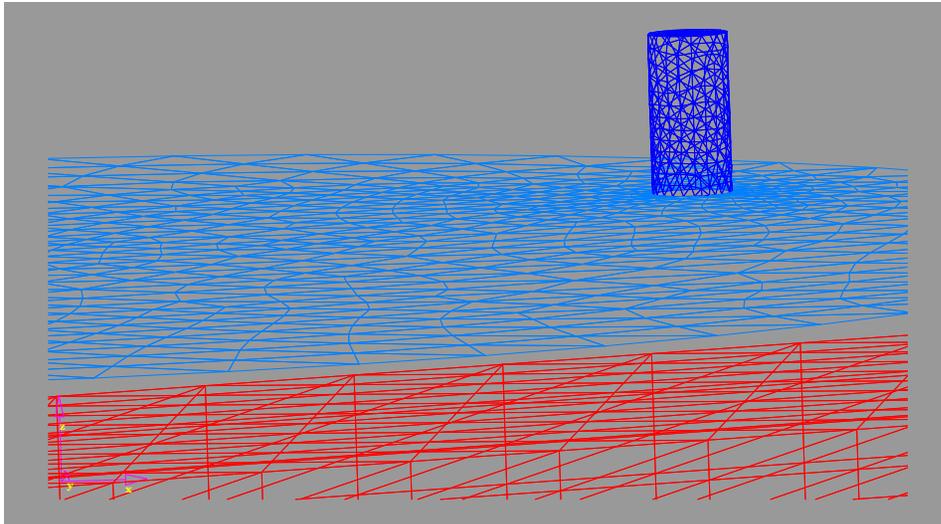
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ARTIFICIAL SHEAR-FREE LIQUID INTERFACE



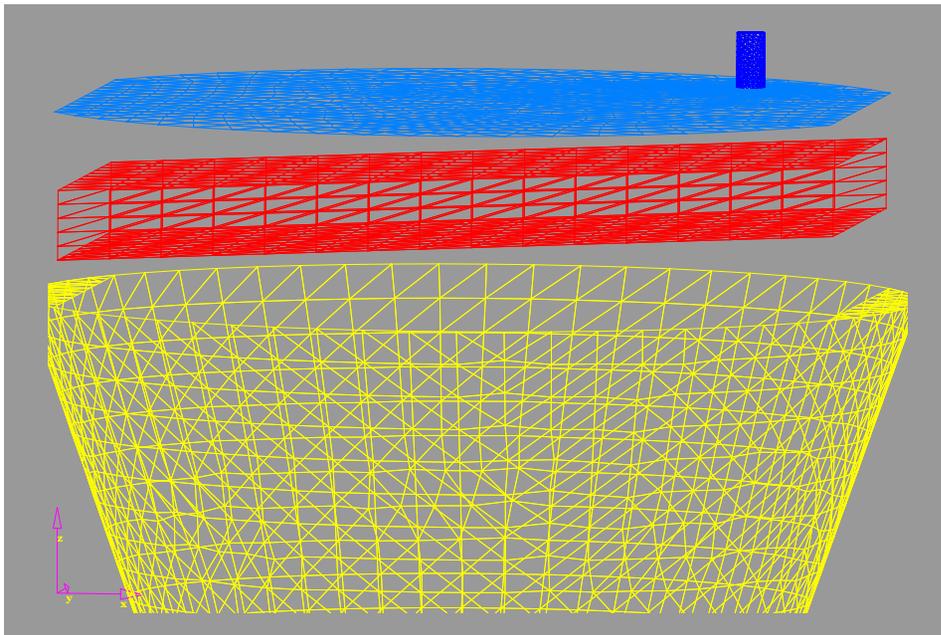
- Upper boundary of the model
- Flat, positioned 6" above the inlet of the tube bundle (East side)
- Intersected perpendicularly by the feed tube (North-East)

FEED TUBE



- 3"-diameter submerged into the liquid (not implemented yet)
- Positioned at the North-East side (B nozzle, 57°)
- Discharge cross section cut at an angle
- Needs additional information (length, orientation of discharge section)
- Needs careful attention to thermal entry length (see B.C.)

TUBE BUNDLE ENVELOPE



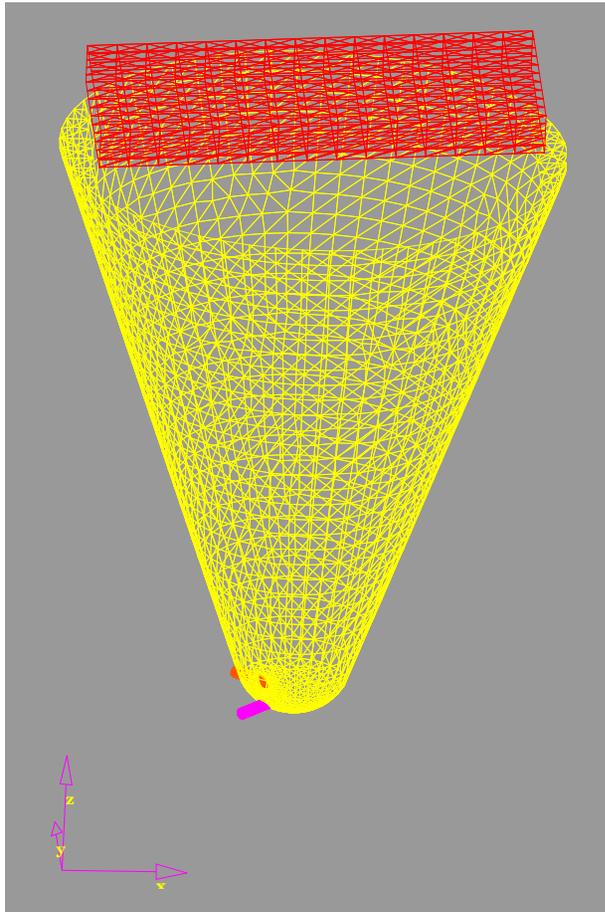
- Duct of rectangular cross section wrapping the tube bundle (240 tubes)
- Traverses the evaporator from East (steam inlet) to West (steam outlet)
- 3.4-ft wide; 7-ft long; 0.625-ft deep;
- Neglects cross flow through bundle
- Individual tubes are 0.758"-diameter; 0.5" gap

TUBE BUNDLE—EAST-WEST VERTICAL CROSS SECTION

(DWG No. D-10760)

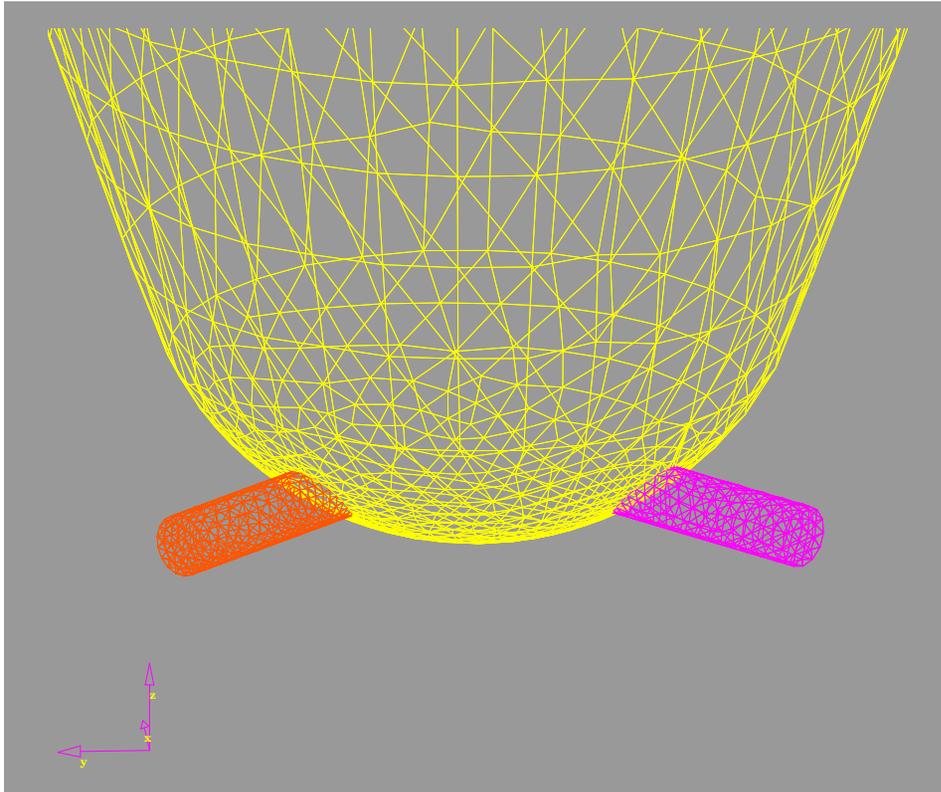
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CYLINDRIC-CONICAL VESSEL



- Cylindric section housing the bundle envelope; 4-ft diameter; 1.75-ft long
- Lower conical section 9-ft long with smallest diameter of 0.84 ft
- Spherical cap 0.5-ft deep at the bottom
- Spherical cap is intersected by the lower lance and discharge lift

DISCHARGE LIFT



- South-West positioning (45°)
- Intersects the spherical cap horizontally 3" from the bottom
- 2"-diameter pipe
- Two discharge lifts shown; only the South-West will be used

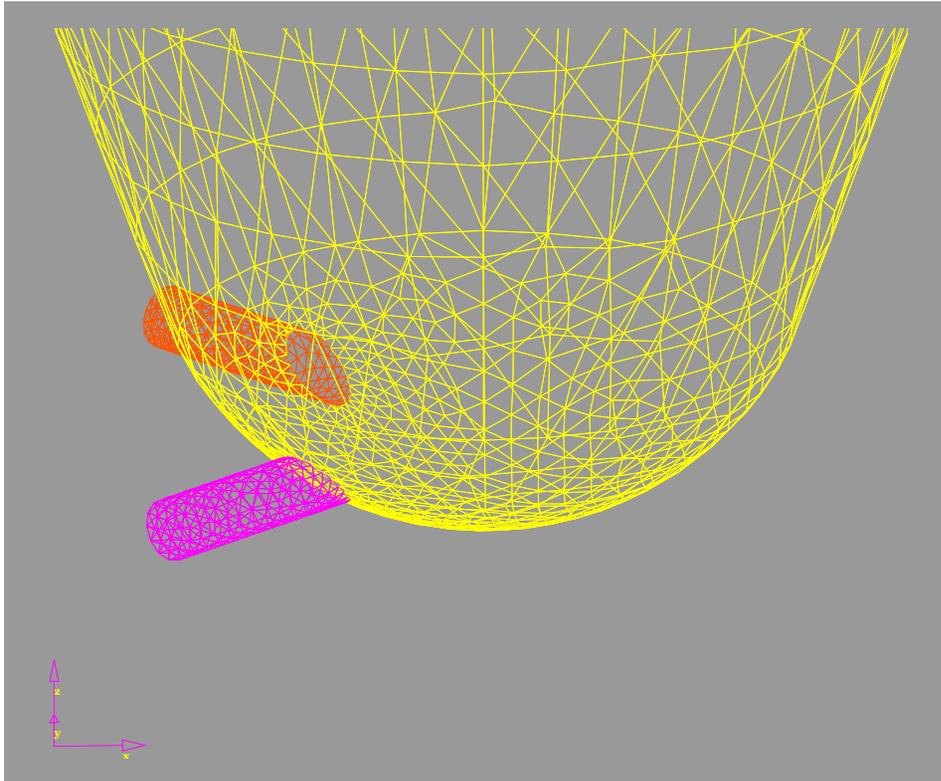
DISCHARGE LIFT Q (DWG No. D-10763)

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DISCHARGE LIFT Q—TOP VIEW (DWG No. D-10763)

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LOWER LANCE



- Positioned South pointing North
- Intersects the spherical cap horizontally 3" from the bottom
- 1.5" -diameter pipe
- Not implemented yet

EQUATIONS OF TRANSPORT (FLOW & HEAT)

$$\operatorname{div}_{\mathbf{x}}(\mathbf{v} \otimes \mathbf{v}) = \operatorname{div}_{\mathbf{x}} \mathbf{T} - Gr \theta \mathbf{g} \quad \forall \mathbf{x} \in \Omega,$$

$$\operatorname{div}_{\mathbf{x}} \mathbf{v} = 0 \quad \forall \mathbf{x} \in \Omega,$$

$$Pr \operatorname{div}_{\mathbf{x}}(\theta \mathbf{v}) = \operatorname{div}_{\mathbf{x}}(\nabla \theta) \quad \forall \mathbf{x} \in \Omega,$$

with constitutive equation of stress

$$\mathbf{T}(\mathbf{x}) := -p(\mathbf{x})\mathbf{I} + \nabla_{\mathbf{x}} \mathbf{v} + \nabla_{\mathbf{x}} \mathbf{v}^{\top} \quad \forall \mathbf{x} \in \Omega.$$

- 3-D, laminar, steady-state flow of a Newtonian incompressible fluid
- Boussinesq buoyancy term coupling bulk fluid flow and heat transport
- Fields: velocity \mathbf{v} , pressure p , and temperature θ

AVERAGE PHYSICAL PROPERTIES OF SIMULANT			
ρ	mass density	1.25	kg m^{-3}
μ	dynamic viscosity	$2.54 \cdot 10^{-3}$	$\text{kg m}^{-1} \text{s}^{-1}$
β	thermal expansion coef.	$5.5 \cdot 10^{-4}$	K^{-1}
C_p	heat capacity	$3.7 \cdot 10^3$	$\text{J kg}^{-3} \text{K}^{-1}$
κ	thermal conductivity	$6.8 \cdot 10^{-1}$	$\text{W m}^{-1} \text{K}^{-1}$

From Bostick and Steele, Thermal and Physical Property Determinations for SRS Waste Simulant Solutions, ORNL/TM-1999/133

FIELDS		UNIT
v	velocity	$\frac{\mu}{\rho D}$
p	pressure	$\frac{\mu^2}{\rho D^2}$
Θ	ref. temperature difference	$T_h - T_f$ ($\Theta = \frac{T - T_f}{T_h - T_f}$)

PARAMETERS & CONSTANTS			
T_f	feed temperature	323–363 (50–90)	(°K)(°C)
T_h	tube bundle temp.	443–463 (170–190)	(°K) (°C)
T_l	lance temperature	443–463 (170–190)	(°K) (°C)
g	gravity	9.81	m s^{-2}
D	ref. diameter	2.4 (8)	m (ft)
\mathcal{Q}_f	feed flow rate	$1.6 \cdot 10^{-3}$ (25)	$\text{m}^3 \text{s}^{-1}$ (gpm)
\mathcal{Q}_d	discharge lift	$6.4 \cdot 10^{-3}$ (10)	$\text{m}^3 \text{s}^{-1}$ (gpm)
\mathcal{Q}_l	lance flow rate	? (?)	$\text{m}^3 \text{s}^{-1}$ (gpm)

DIMENSIONLESS GROUPS		SIGNIFICANCE	RANGE
$Gr := \frac{\rho^2 g D^3 \beta (T_h - T_f)}{\mu^2}$	Grashof	$\frac{\text{buoyance force}}{\text{viscous force}}$	$2.7\text{--}1.5 \cdot 10^6$
$Pr := \frac{C_p \mu}{\kappa}$	Prandtl	$\frac{\text{viscous diffusion}}{\text{heat diffusion}}$	14

- Relatively small Grashof number

BOUNDARY CONDITIONS

- Artificial shear-free liquid interface (surface \mathcal{S}_1)
 - Null traction: $\mathbf{T}\mathbf{n} = \mathbf{0}$ on \mathcal{S}_1 $\mathbf{n} \perp \mathcal{S}_1$
 - * This condition allows flow through the interface
 - No diffusive heat flux: $\partial_n \Theta = 0$ on \mathcal{S}_1
 - * Heat is removed at the interface by convection
- Feed tube (tube wall \mathcal{S}_2)
 - No slip condition: $\mathbf{v} = \mathbf{0}$ on \mathcal{S}_2
 - Constant feed temperature: $\Theta = 0$ on \mathcal{S}_2

- Feed tube (discharge \mathcal{S}_3)

- Specified feed flow rate: $\mathbf{v} = -Re_f \mathbf{i}_z$ on \mathcal{S}_3

- Constant feed temperature: $\Theta = 0$ on \mathcal{S}_3

- Feed tube Reynolds number $Re_f = \frac{4 \rho_f \mathcal{Q}_f}{\pi d \mu_f} = 13$

- Peclet number $Pe = Re Pr = 182$

- Thermal entry length $\approx Pe d / \lambda = 4 \text{ m} \approx 12 \text{ ft}$

- Tube bundle envelope (\mathcal{S}_4)

- No slip condition: $\mathbf{v} = \mathbf{0}$ on \mathcal{S}_4

- Constant temperature: $\Theta = 1$ on \mathcal{S}_4

- Cylindric-conical vessel (discharge \mathcal{S}_5)

- No slip condition: $\mathbf{v} = \mathbf{0}$ on \mathcal{S}_5

- Insulated: $\partial_n \Theta = 0$ on \mathcal{S}_5

- Discharge lift (tube wall \mathcal{S}_6 and discharge outlet \mathcal{S}_7)

- No slip condition: $\mathbf{v} = \mathbf{0}$ on \mathcal{S}_6

- Insulated: $\partial_n \Theta = 0$ on \mathcal{S}_6

- Specified feed flow rate: $\mathbf{v} = Re_d \mathbf{i}_r$ on \mathcal{S}_7

- * Discharge Reynolds number $Re_d = \frac{4 \rho_d \mathcal{Q}_d}{\pi d \mu_d} = 3.5$ (assuming 1.4 specific gravity)

- Fully developed $\partial_n \Theta = 0$ on \mathcal{S}_7

- Lance (\mathcal{S}_8)

- Specified flow rate: $v = -Re_l i_y$ on \mathcal{S}_8

- * Lance Reynolds number $Re_l = \frac{4\rho_l Q_l}{\pi d \mu_l} = ?$

- Specified temperature $\Theta = 1$ on \mathcal{S}_8

→ Boundary conditions add three Reynolds number: Re_f , Re_d , Re_l
(the first two are small)

PARAMETRIC STUDIES & RESULTS

- Compute velocity, temperature and pressure fields for 3 sets of values of five parameters: Gr , Pr , Re_f , Re_d , Re_l
- Visualize the temperature and speed of the flow pointwise
- Visualize trajectories of fluid particles released at the feed and their temperature history

SOLUTION METHOD AND ALGORITHMS

$$Re \nabla_{\mathbf{x}} \mathbf{v} \mathbf{v}(\mathbf{x}) = \operatorname{div}_{\mathbf{x}} \mathbf{T} \quad \forall \mathbf{x} \in \Omega ,$$

$$\operatorname{div}_{\mathbf{x}} \mathbf{v} = 0 \quad \forall \mathbf{x} \in \Omega ,$$

$$\mathbf{v}(\mathbf{x}) = \mathbf{v}_b(\mathbf{x}) \quad \forall \mathbf{x} \in \partial\Omega .$$

$\mathbf{T}(\mathbf{x}) := -p(\mathbf{x})\mathbf{I} + (\nabla_{\mathbf{x}} \mathbf{v} + \nabla_{\mathbf{x}} \mathbf{v}^{\top})$, and natural parameter: Reynolds number Re .
 “Popular” weak form: find $(\mathbf{v}, p) \in \mathbf{V} \times L_0^2(\Omega)$ such that

$$Re \, c(\mathbf{v}, \mathbf{v}, \mathbf{u}) + a(\mathbf{v}, \mathbf{u}) + b(p, \mathbf{u}) = 0 \quad \forall \mathbf{u} \in \mathbf{H}_0^1(\Omega) ,$$

$$b(q, \mathbf{v}) = 0 \quad \forall q \in L_0^2(\Omega) ,$$

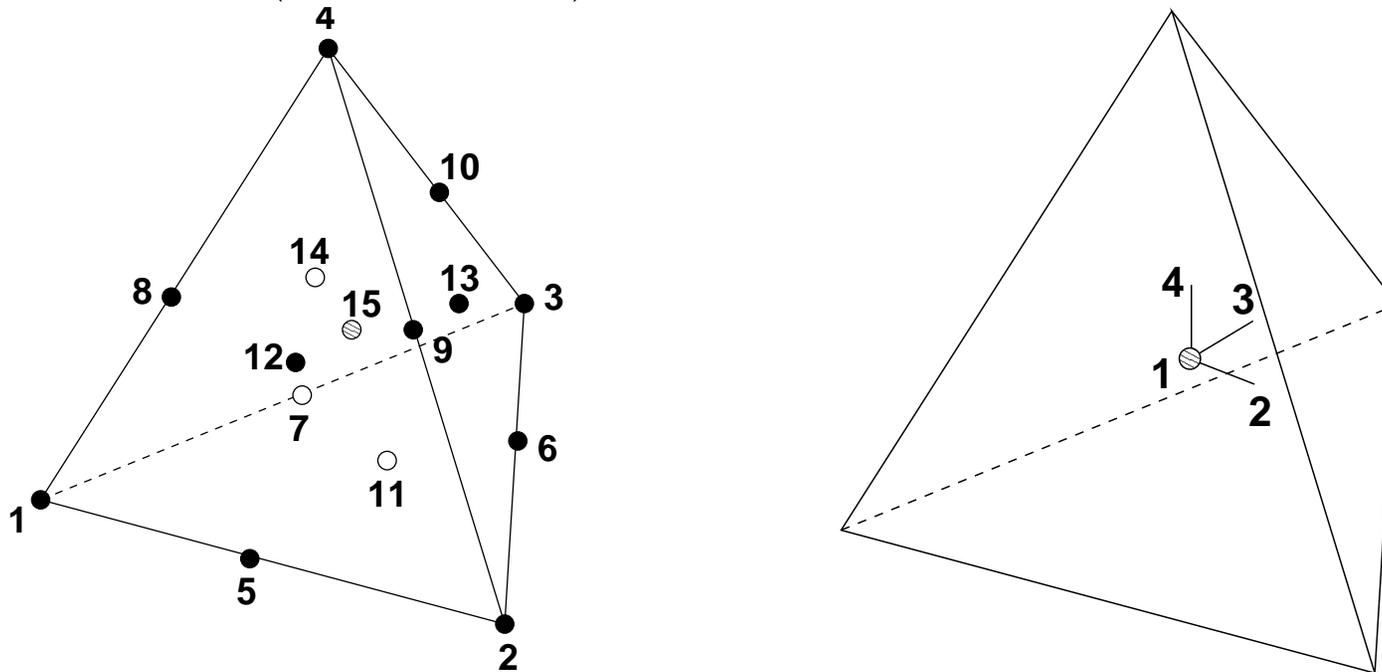
where

$$c(\mathbf{w}, \mathbf{v}, \mathbf{u}) := \int_{\Omega} \nabla_{\mathbf{x}} \mathbf{w} \mathbf{v}(\mathbf{x}) \cdot \mathbf{u}(\mathbf{x}) \, d\mathbf{x} , \quad a(\mathbf{v}, \mathbf{u}) := \int_{\Omega} (\nabla_{\mathbf{x}} \mathbf{v} + \nabla_{\mathbf{x}} \mathbf{v}^{\top}) \bullet \nabla_{\mathbf{x}} \mathbf{u} \, d\mathbf{x} ,$$

$$b(p, \mathbf{u}) := \int_{\Omega} -p(\mathbf{x}) \operatorname{div}_{\mathbf{x}} \mathbf{u} \, d\mathbf{x} , \quad \mathbf{V} := \{ \mathbf{v} \in \mathbf{H}^1(\Omega) \mid \mathbf{v} \equiv \mathbf{v}_b \text{ on } \partial\Omega \} .$$

TETRAHEDRON $P_2^{++}-P_1$

VELOCITY (3×15 d.o.f.) DISCONTINUOUS PRESSURE (4 d.o.f.)

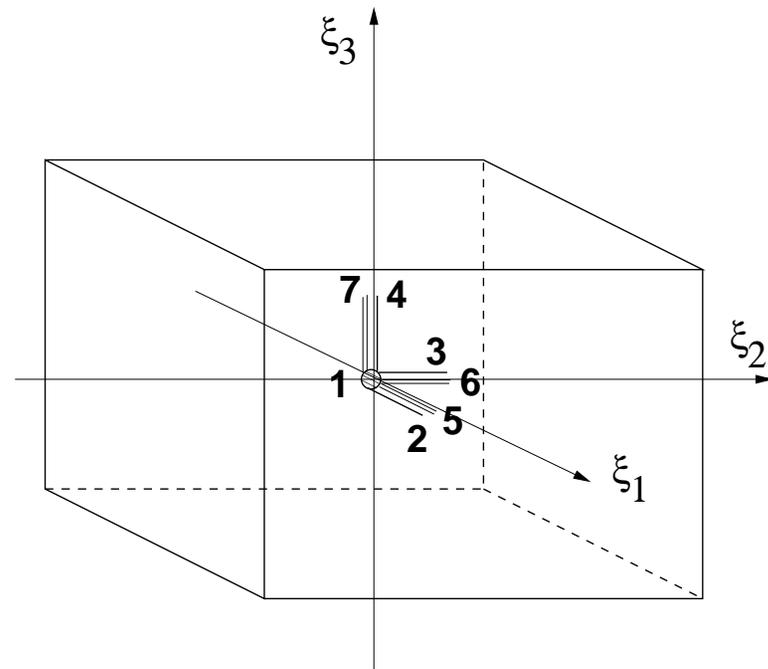
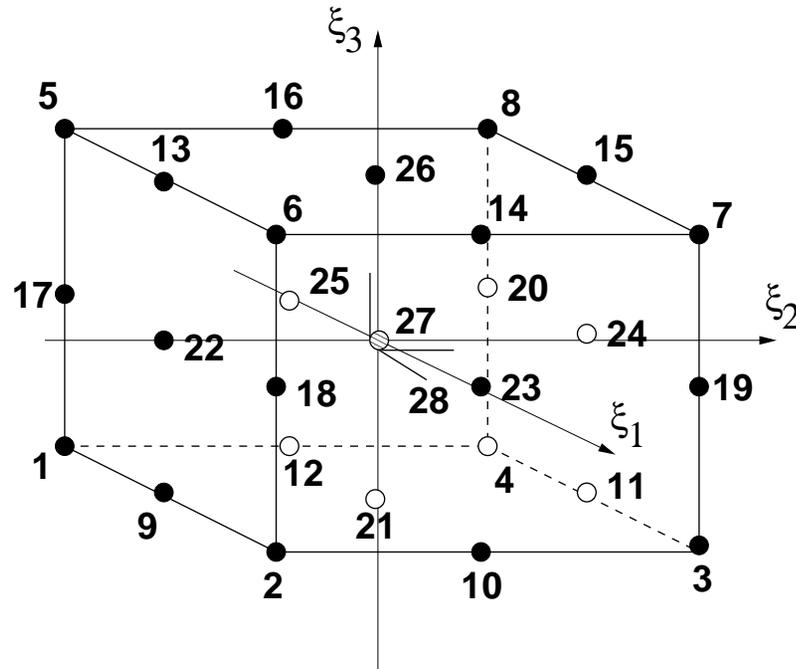


- Based on $P_2^+-P_1$ (Girault and Raviart, 1986)
- Second-order accurate. Stable. Elementwise mass conservation
- Data locality allows static condensation of 7 internal degrees of freedom

HEXAHEDRON $Q_2^+ - P_1^+$

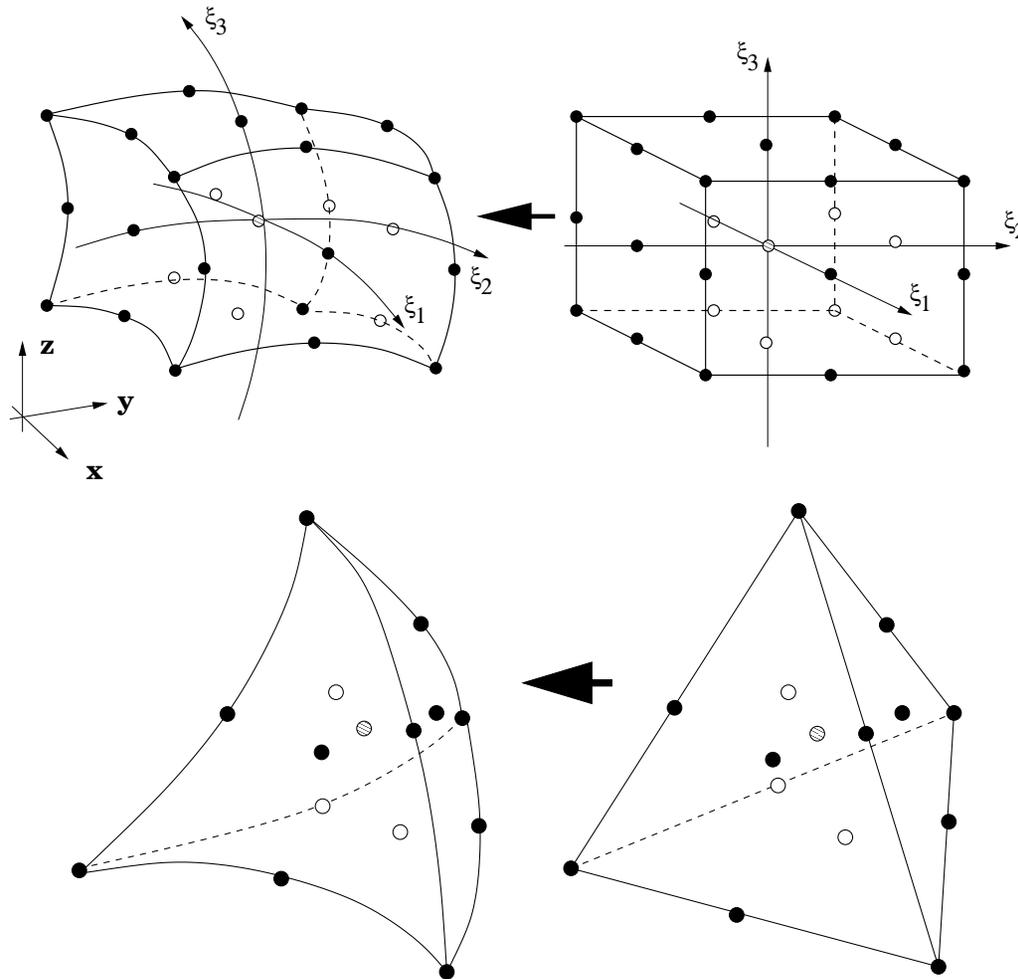
VELOCITY (3×28 d.o.f.)

DISCONTINUOUS PRESSURE (7 d.o.f.)



- Based on $Q_2 - P_1$ (Unknown, 1979)
- Second-order accurate. Stable. Elementwise mass conservation
- Data locality allows static condensation of 13 internal degrees of freedom

ISOPARAMETRIC ELEMENTS

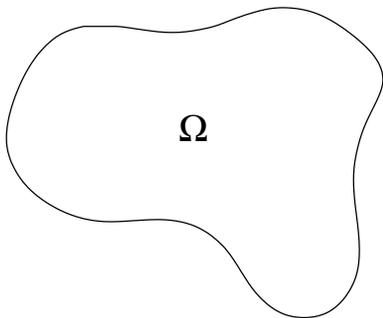


- Can capture geometry curvature with small number of elements

MODIFIED NEWTON-KRYLOV NONLINEAR SOLVER

- Navier-Stokes equations

$$\begin{aligned} Re \nabla \mathbf{v} \mathbf{v} &= \operatorname{div} \mathbf{T} \quad \text{in } \Omega, \\ \operatorname{div} \mathbf{v} &= 0 \quad \text{in } \Omega, \\ \mathbf{v} &= \mathbf{v}_b \quad \text{on } \partial\Omega. \end{aligned}$$



Reynolds number Re .

Stress tensor

$$\mathbf{T} := -p\mathbf{I} + (\nabla \mathbf{v} + \nabla \mathbf{v}^T);$$

- Traditional Newton-like method

- Given $\mathbf{v}^{(0)}$
- Do $n = 1, 2, 3 \dots$ (small)

$$* \text{ Update } \begin{cases} \mathbf{v}^{(n)} \equiv \mathbf{v}^{(n-1)} + \mathbf{v}_c^{(n)}, \\ p^{(n)} \equiv p^{(n-1)} + p_c^{(n)}. \end{cases}$$

- * Solve the linear system

$$\begin{pmatrix} \mathbf{A}^{(n-1)} & \mathbf{B}^T \\ \mathbf{B} & \mathbf{0} \end{pmatrix} \begin{pmatrix} \mathbf{v}_c^{(n)} \\ p_c^{(n)} \end{pmatrix} = \begin{pmatrix} \mathbf{b}_v^{(n-1)} \\ \mathbf{b}_p^{(n-1)} \end{pmatrix}$$

by factorization.

- EndDo

- Shamanskii's modification (inner iterations)

- For a given outer iteration n
- Do $k = 1, 2, 3 \dots$

* Update
$$\begin{cases} \mathbf{v}^{(k)} \equiv \mathbf{v}^{(k-1)} + \mathbf{v}_c^{(k)}, \\ p^{(k)} \equiv p^{(k-1)} + p_c^{(k)}. \end{cases}$$

* Solve “approximately” the linear system

$$\begin{pmatrix} \mathbf{A}^{(n-1)} & \mathbf{B}^T \\ \mathbf{B} & \mathbf{0} \end{pmatrix} \begin{pmatrix} \mathbf{v}_c^{(k)} \\ p_c^{(k)} \end{pmatrix} = \begin{pmatrix} \mathbf{b}_v^{(k-1)} \\ \mathbf{b}_p^{(k-1)} \end{pmatrix}$$

with a preconditioned iterative solver

- EndDo

- ILUTF($u, tol, lfil$) GMRES

- frontal-based incomplete factorization
- u threshold pivoting
- tol numerical dropping
- $lfil$ fill-in cut-off point

OUTLOOK

Suggested improvements towards more realistic modeling

- Parallel computing (possibly terascale)
 - Improved geometry
 - Finer discretization
 - Wider range of parameter study

- Incorporate
 1. Multicomponents
 2. Chemical reactions
 3. Two-phase model