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Development and Characterization of an Improved Ir-0.3% W Alloy for Space Radioisotopic Heat Sources

C. T. Liu
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DEVELOPMENT AND CHARACTERIZATION OF AN IMPROVED Ir-0.3% W ALLOY
FOR SPACE RADIOISOTOPIC HEAT SOURCES

C. T. Liu and H. Inouye

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DEVELOPMENT AND CHARACTERIZATION OF AN IMPROVED Ir-0.3% W ALLOY
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C. T. Liu and H. Inouye

ABSTRACT

An Ir-0.3% W alloy has been doped with minor alloying additions to develop an improved cladding material for space radioisotopic heat sources operating at temperatures to 1450°C. Tensile tests of a series of doped alloys at slow strain rates indicated that the DOP-4 alloy containing 40 Al, 30 Th, 80 Fe, 10 Ni, and 75 Rh ppm was most resistant to the brittle fracture associated with grain-boundary separation. In addition, the DOP-4 dopants raise the recrystallization temperature, and retard grain growth at high temperatures.

The impact properties of DOP-4 and undoped Ir-0.3 W alloys were determined as a function of grain size, test temperature, impact velocity and long-term heat treatment. All the results indicate that the impact properties of the DOP-4 alloy are far superior to those of the undoped alloy. The improvement results from the segregation of beneficial dopant(s) (such as thorium) to grain boundaries and precipitation of second-phase particles. The development of the DOP-4 alloy greatly improves the safety margin of the Multi-Hundred-Watt heat sources to be used in a space probe to Jupiter and Saturn in 1977.

INTRODUCTION

A multi-hundred watt (MHW) radioisotope heat source for a thermo-electric generator (RTG) has been developed by General Electric (GE) under the sponsorship of the Space Nuclear Systems Division¹ of ERDA to provide stable electrical power for a variety of space missions (Fig. 1). The heat source² is 18.3 cm in diameter and 42.9 cm long, and weighs about 20.4 kg.

¹Presently part of the Division of Nuclear Research and Applications.

²General Electric Company, *Multi-Hundred Watt Radioisotopic Thermo-Electric Generator Program, Parts I and II, Annu. Rep. 1 Jan. 1973-31 Dec. 1973*, GESP-7107, GEMS-418.

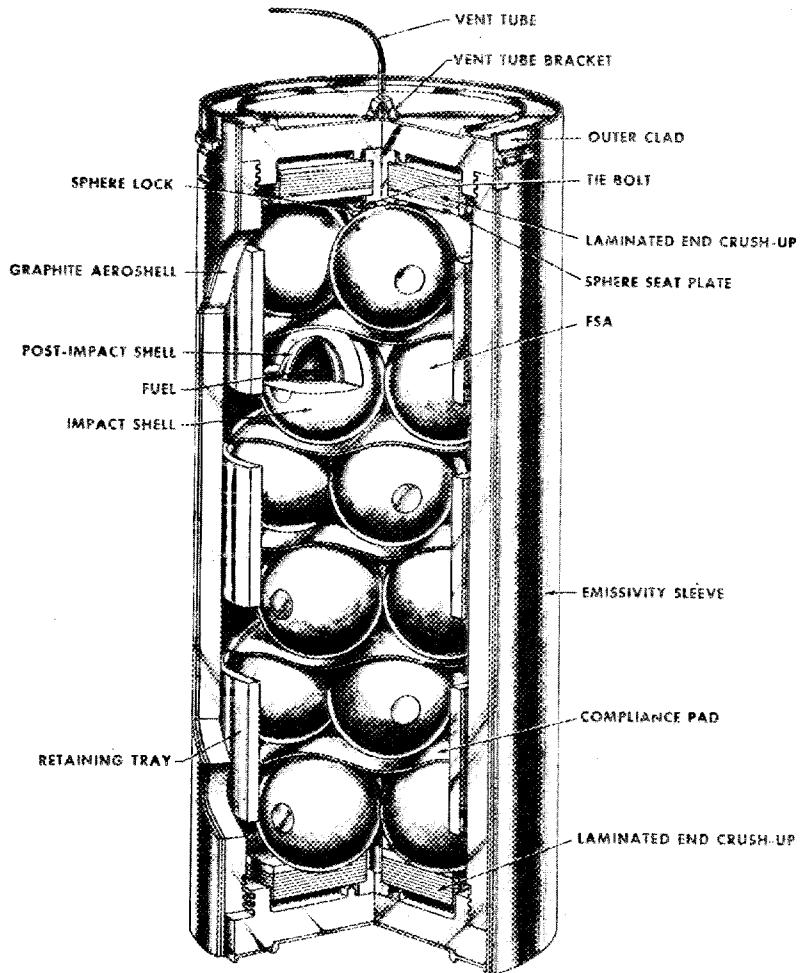


Fig. 1. Multihundred-Watt Heat Source Designed by General Electric Company. Reprinted from General Electric Company, *Multihundred Watt Radioisotopic Thermo-Electric Generator Program, Part I, Annu. Rep. 1 Jan. 1973-31 Dec. 1973*, GESP-7107.

It is fueled with 24 fuel sphere assemblies (FSA). The FSA (Fig. 1) has three parts: (1) a 3.81-cm-diam $^{238}\text{PuO}_2$ fuel ball, (2) a 0.051-cm-thick postimpact containment shell (PICS) made of oxidation-resistant alloy, and (3) a 1.0-cm-thick thornel graphite impact shell (GIS). The fuel-cladding is designed for an operating temperature around 1330°C. The electrical power is generated by silicon-Germanium thermoelectric elements attached to the outer can of the heat source. An alloy of Ir-0.3% W is currently used as PICS material in the MHW heat source because of the alloy's high melting point, good oxidation

resistance and proven compatibility with fuel and graphite to 1450°C. Tungsten, at a level of 0.3%, is added to increase the recrystallization temperature and improve the fabricability of iridium. A detailed description of the production of Ir-0.3% W sheets and disks for fuel cladding in the MHW heat source has been reported.³

On tensile testing at conventional strain rates (10^{-4} m/s), the Ir-0.3% W alloy has been found to be ductile at temperatures above 1000°C.⁴ The alloy specimens had more than 50% tensile elongation and fractured by ductile rupture with close to 100% reduction of area. To simulate the accident conditions of earth impact after orbital abort and atmosphere reentry situation, the FSA/MHW have been impact tested at 1370°C at a velocity of 90 m/s (300 fps) at GE and Los Alamos Scientific Lab (LASL). Their results indicates^{2,5} that the Ir-0.3% W PICS sometimes survive the impact test but quite often showed brittle cracking with limited deformation even at a temperature of 1450°C. In the more severe cases, the cracking leads to a breach of the containment shell which could release small quantities of radioactive fuel. Because of the marginal impact resistance of the Ir-0.3% W, there is a need to improve the impact properties of the alloy to increase the safety margin of the space heat sources.

Posttest examination of iridium-tungsten PICS revealed^{2,5,6} that the fracture mode of these cracks was mainly intergranular on impact at 90 m/s at 1400°C. The same fracture mode was also observed⁴ at slow strain-rate tests at temperatures below 800°C. Surface analyses by Auger and spark-source mass spectrometry indicated no major segregation of impurities on the grain boundary,

³D. N. Braski and A. C. Schaffhauser, *Production of Ir-0.3% W Disks and Foil*, ORNL-TM-4865 (April 1975).

⁴C. T. Liu and H. Inouye, *Study of Iridium and Iridium-Tungsten Alloys for Space Radioisotopic Heat Sources*, ORNL-5240 (December 1976).

⁵R. D. Baker, Los Alamos Scientific Lab., private communications during 1974-1976.

⁶C. T. Liu and H. Inouye, *Postmortem Analysis of Ir-2% W Post-Impact Containment Shell*, ORNL-TM-4943 (July 1975).

suggesting that the grain-boundaries in these high-purity iridium and Ir-0.3% W alloys are intrinsically weak. The purpose of this present work was to improve the impact properties of Ir-3% W alloy through control of the grain boundary chemistry and minor alloying elements (without sacrificing the other desirable properties).

In this study, selected alloying additions (dopants) in the ppm (by weight) range were added to the Ir-0.3% W alloy. To characterize the impact properties of the doped alloys, impact equipment has been developed at ORNL, which is capable of testing sheet specimens in controlled atmospheres at 1400°C at velocities to 90 m/s (300 fps).

SCREENING STUDY OF DOPED Ir-0.3% W ALLOYS

Selection of Alloying Elements (Dopants)

The selection of dopants for this study was based on the following considerations:

- a. Previous study⁴ of tensile fracture behavior of Ir and Ir-W alloys revealed that alloys containing higher levels of Th, Al, Fe, Ta, Ni, and Rh were more resistant to grain boundary fracture. Furthermore, the concentrations of Th, Ta, Fe, and Al were much higher near the grain-boundary than in the bulk, suggesting the possibility of segregation of these elements to the grain boundary. These elements may act synergistically to facilitate grain boundary segregation, so they were added as a group to the Ir-0.3% W alloy.
- b. Some early work⁷⁻⁹ on iridium polycrystals suggested that the brittleness in iridium is due to segregation of interstitials on the grain boundaries. If it is the case, alloying iridium with an active component, such as hafnium, may scavenge interstitials from the boundary as well as from the matrix by precipitation.

⁷M. A. Fortes and B. Ralph, "A Field-Iron Microscope Study of Segregation to Grain Boundaries In Iridium," *Acta. Met.* 15: 707 (1976).

⁸Personal communication from P. W. Palmbert to R. L. Mehan, General Electric Company, Philadelphia, December, 9, 1971.

⁹C. A. Brookes, J. H. Greenwood, and J. L. Routbort, "The High-Temperature Tensile Properties of Iridium Single Crystals," *J. Inst. Met.* 98: 27 (1970).

- c. Yttrium is the most common element used for improving grain-boundary properties in metals and alloys.

Alloy Preparation and Fabrication

The iridium powders (lots WC and WG) used to prepare doped Ir-0.3% W alloys were obtained from U.S. Office of Emergency Preparedness. The chemical analysis by spark source mass spectrometric methods indicated³ that these powders were quite pure with total impurities less than 300 ppm by wt. The major impurities were Ca, Si, Cl, and Rh; most of the volatile impurities are removed during melting. Two series of doped alloys were prepared with nominal concentrations of dopants (Table 1). The tungsten and dopants (except for hafnium and yttrium) in powder form were thoroughly mixed with iridium powders, followed by cold pressing into compacts. The compacts were sintered in hydrogen at 1000°C for one hour and vacuum sintered at 1300°C for 3 hr. The sintered compacts were then arc-melted six times and cast into rectangular-shaped ingots about $0.8 \times 2.5 \times 5.1$ cm, weighing about 150 g. For the doped alloys DOP-11 and -12, the hafnium and yttrium lumps were added to the Ir-0.3% W during arc-melting. A few white

Table 1. Nominal Concentration of Dopants in Ir-0.3% W Alloys

ceramic-like globules were observed on the top surface of the ingot DOP-8, but not on the rest of the ingots. The globules are most probably complex oxides and were insoluble in hot NaOH, aqua regia, and hafnium solutions. One half of ingot DOP-10 was badly contaminated with copper during casting, and so was not used. Alloys DOP-7 and -8 contain double and triple concentrations of the dopants in DOP-4 (referred to as DOP-4 dopants). DOP-10 contains 170 ppm Th plus DOP-4 dopants.

The as-cast alloy ingots were first cleaned in a solution of aqua regia plus 15% HF solution for about 30 min., and were next clad in molybdenum jackets and sealed by electron-beam welding. The assemblies were then heated at 1200–1250°C under a partial argon atmosphere and rolled with 15–18% reduction per pass and a 10-min-reheating period. After a total of 65% reduction, the alloy plates (~0.25-cm-thick) were removed from the molybdenum jacket and cleaned electrolytically in KCN solution. After recrystallizing for 1 hr at 1300–1400°C the alloy plates were wrapped in loose-fitting molybdenum cover sheet and further rolled at 900 to 1100°C to 0.6-mm-thick sheets. The finished sheets generally exhibited some minor edge and end cracks, except for DOP-11 which showed severe edge and surface cracks. Microscopic examination indicates that the surface cracks on sheet DOP-11 were formed along grain boundaries. This observation suggests that alloying Ir–0.3% W with 150 ppm Y has a detrimental effect on its grain boundary ductility and fabrication properties. Thus, the DOP-11 sheet was not used for further evaluations.

Chemical compositions of the doped alloys as determined by the SSMS analysis show that the dopant levels correlate reasonably well with the nominal concentrations in the alloys prepared by arc-melting (Table 2). The Al, Th, and Fe contents are lower than the nominal amount, possibly due to loss through evaporation or formation of glassy globules (as observed on the ingot DOP-8). The alloy DOP-8 picked up a significant amount of iron, and DOP-10 iron and copper during arc melting and casting. These alloys had a low level of

Table 2. Chemical Analysis^a of Doped Ir-0.3% W Alloy Sheets

Element	First Series of Doped Alloys ^b					Second Series of Doped Alloys ^b				Scaled-up Alloy ^c	Undoped Alloy ^c
	DOP-1	DOP-2	DOP-3	DOP-4	DOP-5	DOP-7	DOP-8	DOP-10	DOP-12	DOP-4-1	WG
Al ^d	20	10	20	20	20	70	100	100	10	10	5
B	3	0.1	0.1	0.1	0.1	<0.1	<0.1	<0.1	0.1	0.5	0.5
Ca	0.3	0.5	0.1	0.1	0.6	1	0.3	10	1	1	1
Co	1	1	1	1	1	1	1	1	1	<0.3	0.1
Cr	5	3	15	15	5	20	3	3	3	<3	55
Cu	1	7	7	7	20	15	15	150	50	20	1
Fe ^d	50	50	10	50	50	150	600	200	100	15	5
Hf ^d	<0.3	<0.3	<0.3	<0.3	<0.3	<0.3	<0.3	<0.3	120	<0.3	<0.3
Mn	0.3	0.5	0.5	0.5	0.5	1	3	3	0.5	0.3	0.1
Ni ^d	20	10	20	20	20	50	60	20	5	5	1
P	0.1	0.1	0.1	0.1	0.3	0.3	0.3	0.3	<0.1		
Pt ^d	15	10	20	30	15	10	30	20	20		
Rh ^d	80	80	80	100	100	100	200	150	10	70	20
Ru	15	15	15	15	15	1	10	10	20	50	20
Si	15	15	15	40	15	3	100	30	10	5	5
Ta ^d	20	30	20	10	40	50	20	20	10	10	10
Th ^d	15	20	15	15	<0.5	50	60	120	<1	20	<0.1
W ^e											
Y ^d	<0.1	<0.1	<0.1	<0.1	<0.1	<0.1	<0.1	<0.1	<0.1	<0.1	<0.1

^a Analysis in parts per million by spark source mass spectrometric method.^b Alloys were prepared by arc melting.^c Alloys were prepared by electron-beam melting.^d Dopant elements with nominal concentration.^e Assumed to be 30.00 ppm in all material.

interstitial impurities (Table 3) and were not significantly different from the electron-beam melted, undoped Ir-0.3% W alloy.

Microstructure and Recrystallization

The doped alloy sheets produced by warm-rolling between 900-1100°C had a fibrous microstructure. Occasionally, a few small recrystallized grains were observed. The sheet specimens of DOP-1 and -4 were vacuum heat-treated for 1 hr at temperatures between 800 and 1600°C to study their recrystallization and microstructure. The recrystallization temperatures were determined metallographically (Table 4) and compared with those for undoped Ir-0.3% W. The recrystallization of the undoped alloy started at 870°C and was complete at 1070°C for 1 hr heat treatments. However, the recrystallization temperatures of the two alloys with a total of 240-270 ppm dopants were 150°C higher. Thus, the recrystallization behavior of Ir-0.3% W is sensitive to the trace elements in the parts per million range.

The recrystallized doped alloys had a few second-phase particles distributed uniformly throughout the grains. The doped alloys had an elongated grain structure when recrystallized which became equiaxed after heat treatments above 1500°C.

Table 3. Interstitial Content of Doped Ir-0.3% W Alloys

Alloy	Interstitial Content, ^a ppm by weight			
	C	H	N	O
DOP-1 ^b	6	3	1	14
DOP-4 ^b	8	2	3	10
Undoped ^c	9	1	1	5

^aDetermined by vacuum fusion analysis.

^bArc-melted,

^cElectron-beam melted.

Table 4. Effect of 1 hr Heat-Treatment on the Recrystallization of DOP-1, DOP-4,
and Undoped Ir-0.3% W Alloys^a

Alloy	Temperature for Indicated Percentage of Recrystallization, °C	
	0% ^b	100% ^b
DOP-1	1000	1230
DOP-4	1000	1230
Undoped	870	1070

^a Warm rolled between 900–1100°C.

^b Percent of recrystallization.

Tensile Properties

Specimens with a gage section of 0.318 cm wide by 1.3 cm long and with a 0.310-cm-diam pin hole were blanked from 0.05–0.08-cm-thick stock for tensile testing. The blanked specimens were polished on 400-A SiC paper, pickled in an aqua regia plus HF solution, then heat-treated for 1 hr at 1500°C in vacuum. Tensile specimens tested above room temperature were heated by radiation from an inductively heated tantalum susceptor under a vacuum of less than 1×10^{-3} Pa. A Pt vs Pt–10% Rh thermocouple centrally located in the specimen monitored the temperature. After a 15-min holding time at the test temperature, tests were made on an Instron testing machine at a strain rate of 0.26–0.51 cm/min (0.1–0.2 in./min).

The tensile properties of the first series of doped alloys were measured at room temperatures, 650, 760, 1093, and 1370°C (Table 5); and compared with undoped Ir-0.3% W alloy (lots WG and WC) prepared from electron-beam melted and drop-cast ingots. The yield strength of the doped alloys DOP-1 to -5 is higher than that of the undoped alloy by about 50% at all test temperatures. Alloys DOP-1 to -4 had the same level of tensile strength and elongation as the undoped alloy. The elongation of DOP-5 containing no thorium was lower than the alloys doped with 30 ppm Th.

Table 5. Tensile Properties^a of the First Series of Ir-0.3% W Alloys^b

Alloy	Strength, MPa (ksi)		Elongation (%)	Fracture Mode ^c
	Yield	Tensile		
<u>Room Temperature</u>				
Undoped (WC) ^d	84.8 (12.3)	400.3 (58.1)	12.9	Mainly GBS
Undoped (WG) ^d	81.3 (11.8)	431.3 (62.6)	14.2	Mainly GBS
DOP-1 ^e	121.2 (17.6)	427.9 (62.1)	12.2	Mainly GBS
-2 ^e	103.4 (15.0)	347.3 (50.4)	10.4	Mainly GBS
-3 ^e	110.2 (16.0)	416.9 (60.5)	11.6	Mainly GBS
-4 ^e	120.6 (17.5)	445.1 (64.6)	13.1	Mainly GBS
-5 ^e	117.8 (17.1)	381.7 (55.4)	10.3	Mainly GBS
<u>650°C</u>				
Undoped (WC)	65.5 (9.5)	414.1 (60.1)	23.3	Mainly GBS
Undoped (WG)	51.0 (7.4)	487.8 (70.8)	30.1	Mainly GBS
DOP-1	97.1 (14.1)	492.6 (71.5)	29.4	TF(Ma) and GBS (Mi)
-2	79.9 (11.6)	458.9 (66.6)	25.9	TF(Ma) and GBS (Mi)
-3	83.4 (12.1)	471.9 (68.5)	27.6	TF(Ma) and GBS (Mi)
-4	96.5 (14.0)	462.3 (67.1)	31.7	Mainly TF
-5	96.5 (14.0)	436.1 (63.3)	22.2	TF and GBS
<u>760°C</u>				
Undoped (WC)	61.3 (8.9)	445.1 (64.6)	39.8	GBS and TF
Undoped (WG)	46.9 (6.8)	441.0 (64.0)	39.3	GBS (Ma) and TF (Mi)
DOP-1	106.1 (15.4)	450.6 (65.4)	37.9	TF
-2	94.4 (13.7)	423.7 (61.5)	38.2	TF
-3	95.1 (13.8)	425.1 (61.7)	33.1	TF
-4	100.6 (14.6)	425.8 (61.8)	35.2	TF
-5	83.4 (12.1)	414.1 (60.1)	27.3	Mainly TF
<u>1093°C</u>				
Undoped (WC)	44.8 (6.5)	276.3 (40.1)	57.6	DR
Undoped (WG)	46.2 (6.7)	249.4 (36.2)	55.6	DR
DOP-1	63.4 (9.2)	281.1 (40.8)	49.4	DR
-2	58.6 (8.5)	275.6 (40.0)	45.2	DR
-3	68.2 (9.9)	261.8 (38.0)	52.9	DR
-4	69.6 (10.1)	254.9 (37.0)	48.2	DR
-5	62.7 (9.1)	260.4 (37.8)	42.3	DR
<u>1370°C</u>				
Undoped (WC)	34.5 (5.0)	181.2 (26.3)	54.2	DR
Undoped (WG)	38.6 (5.6)	177.8 (25.8)	55.2	DR
DOP-1	73.0 (10.6)	190.9 (27.7)	54.4	DR
-2	50.3 (7.3)	185.3 (26.9)	52.9	DR
-3	51.0 (7.4)	181.9 (26.4)	56.0	DR
-4	53.1 (7.7)	181.9 (26.4)	52.2	DR
-5	56.5 (8.2)	177.1 (25.7)	45.0	DR

^aTested at a crosshead speed of 0.05 to 0.2 in./min in vacuum.^bSheet specimens were annealed 1 hr at 1500°C before testing.^cGBS = grain-boundary separation, TF = transgranular fracture, DR = ductile rupture, Mi = minor fraction, and Ma = major fraction.^dUndoped alloy prepared by electron-beam melting and drop casting.^eDoped alloy prepared by arc melting and regular casting.

Fracture surfaces were examined by optical and/or scanning electron microscopy. The doped and undoped alloys had primarily grain-boundary separation (GBS) at room temperature; however, the doped alloys had more transgranular fracture (TF). At 650°C, GBS was still the dominant mode of fracture for the undoped alloy [Fig. 2(a)], while mainly TF or mixture of TF (major) and GBS (minor) was the fracture mode for the doped alloys. The DOP-4 (containing no tantalum) is the only doped alloy showing mainly TF fracture at 650°C [Fig. 2(b)]. At 760°C, the undoped alloy showed a mixture of

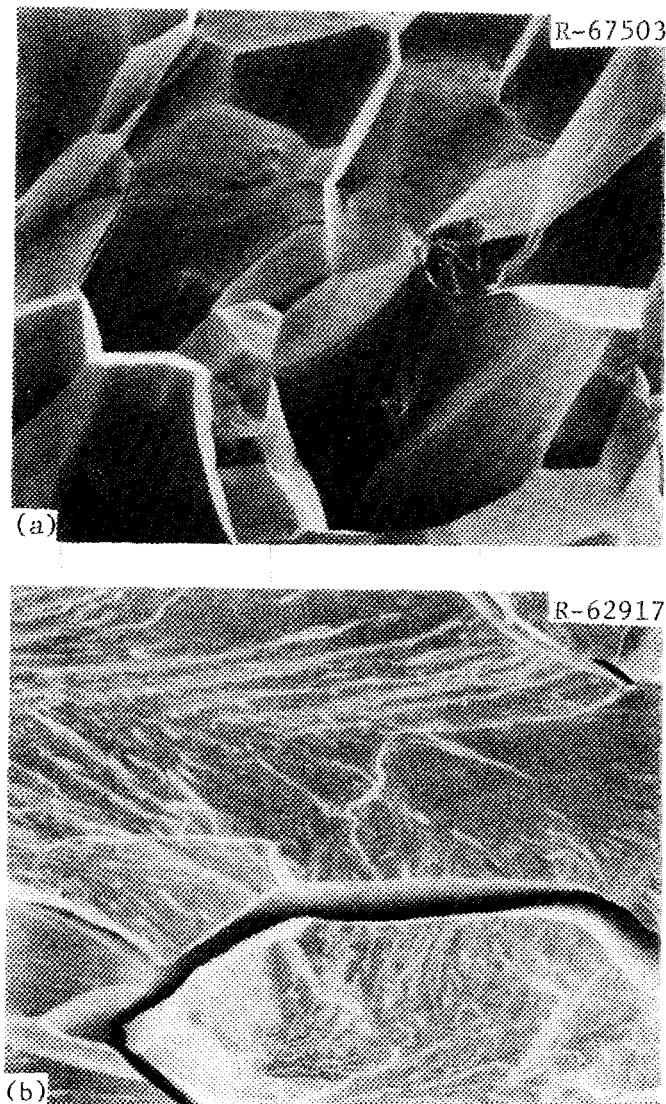


Fig. 2. SEM Fractograph of Tensile Specimens Fractured at 650°C; (a) Undoped WG, and (b) DOP-4. 300×.

GBS and TF, while the doped alloys DOP-1 to -4 exhibited almost complete TF. The intergranular fracture was not observed at higher temperatures; in fact, all alloys, whether doped or not, exhibited ductile rupture with reduction in area close to 100% at 1093 and 1370°C.

Four important correlations can be made between the fracture behavior and dopant composition (Table 5):

1. Doping of Ir-0.3% W with small amounts of Al, Fe, Ta, Th, Ni and Rh at a level less than 100 ppm apparently affects the mechanical properties of the grain boundary and suppresses intergranular fracture at lower temperatures.
2. Test results (Table 5) suggest that several elements rather than just one may contribute to the overall change in the fracture mode.
3. The element which most effectively suppresses GBS in Ir-0.3% is thorium, while the element which may have a detrimental effect is tantalum.
4. The alloy with most resistance to GBS is DOP-4 which was doped with 40 ppm Al, 80 Fe, 30 Th, 16 Ni, and 75 Rh.

To determine the optimum amount of DOP-4 dopants, a second series of doped alloys was prepared (Table 1). DOP-7 and -8 contained double and triple the amounts of DOP-4 dopants, respectively. The nominal amount of thorium (the most beneficial dopant) was increased in DOP-10 (200 ppm) compared to DOP-4 (30 ppm). Unfortunately the DOP-10 alloy was contaminated with copper and iron during arc-melting. The fracture behavior (in terms of percent of TF) of DOP-7, -8, -10, and -12 (doped with 200 ppm Hf alone) can be compared with that of DOP-4 and the undoped alloy at 650 and 760°C, the temperatures where the fracture mode is most sensitive (Fig. 3). The DOP-4 alloy is most resistant to intergranular fracture at these two temperatures (Fig. 3). Thus, the optimum concentration of dopants should be somewhere around the DOP-4 levels. The fracture behavior of DOP-12 is not different from the undoped alloy, indicating that doping with 200 ppm Hf does not improve the fracture behavior of Ir-0.3% W alloy.

Table 6. Tensile Properties of DOP-4 and Undoped Ir-0.3% W Alloys^a
Fabricated from the Same Iridium Powder (WG)

Heat	Strength, MPa (ksi)		Elongation (%)	Fracture Mode ^b
	Yield	Tensile		
<u>Room Temperature</u>				
WG-202 ^c	81.3 (11.8)	431.3 (62.6)	14.2	Mainly GBS
DOP-4-1 ^d	108.9 (15.8)	500.2 (72.6)	15.3	Mainly GBS
DOP-4-3 ^e	95.1 (13.8)	485.1 (70.4)	15.6	Mainly GBS
<u>650°C</u>				
WG-202	51.0 (7.4)	487.8 (70.8)	30.1	Mainly GBS
DOP-4-1	68.9 (10.0)	553.3 (80.3)	36.0	GBS (Ma) and TF (Mi)
DOP-4-3	64.1 (9.3)	542.9 (78.8)	35.7	GBS (Ma) and TF (Mi)
<u>760°C</u>				
WG-202	46.9 (6.8)	441.0 (64.0)	39.3	GBS (Ma) and TF (Mi)
DOP-4-1	60.6 (8.8)	474.0 (68.8)	50.0	TF
DOP-4-3	59.3 (8.6)	465.1 (67.5)	47.1	TF
<u>1093°C</u>				
WG-202	46.2 (6.7)	249.4 (36.2)	55.6	DR
DOP-4-1	55.8 (8.1)	280.4 (40.7)	58.2	DR
DOP-4-3	59.9 (8.7)	268.0 (38.9)	58.8	DR
<u>1370°C</u>				
WG-202	38.6 (5.6)	177.8 (25.8)	55.2	DR
DOP-4-1	46.2 (6.7)	191.5 (27.8)	60.7	DR
DOP-4-3	40.7 (5.9)	187.4 (27.2)	58.1	DR

^aSpecimens annealed 1 hr at 1500°C.

^bGBS = grain boundary separation; TF = transgranular fracture; Ma = major fraction; Mi = minor fraction; and DR = Ductile rupture.

^cUndoped Ir-0.3% W prepared from electron-beam melting and drop casting of WG powder compacts.

^dDoped Ir-0.3% W prepared from electron-beam melting and drop casting of WG powder compacts with DOP-4 additions.

^eDoped Ir-0.3% W prepared from electron-beam melting and drop casting of recycled heads and skulls from ingots, DOP-4, -1, and -2.

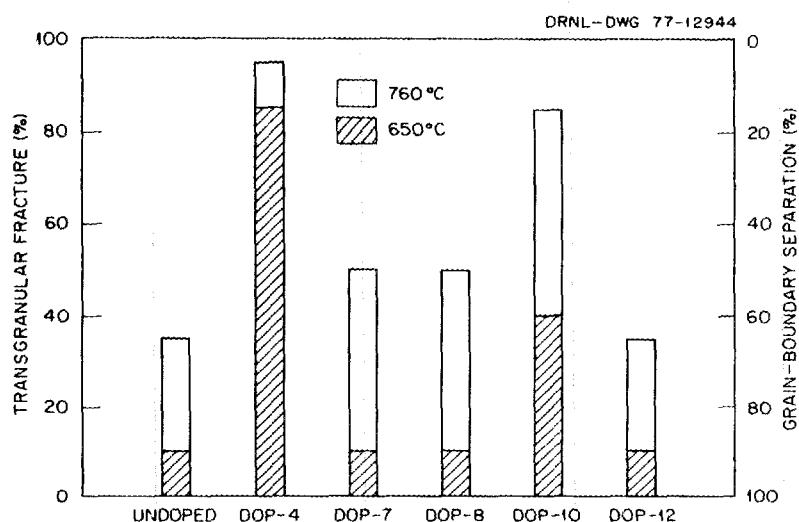


Fig. 3. Fracture Behavior of Doped and Undoped Alloys Tensile Tested at 650 and 760°C at Slow Strain Rate (1×10^{-5} m/s).

SCALE-UP STUDY OF DOP-4 Ir-0.3% W

The screening study indicated that DOP-4 alloy was most resistant to the brittle grain-boundary fracture, so this alloy was chosen for a scale-up study. The purpose of this study was to further evaluate the alloy's fabricability, metallurgical properties (such as microstructure and grain-growth behavior), tensile properties, and impact behavior. The latter evaluation involved an uniaxial impact test developed at ORNL as well as the FSA-Type impact test required for qualification of the MHW heat source.

Alloy buttons weighing 500 g were prepared by both arc melting and electron-beam melting of Ir-0.3% W sintered compacts with the DOP-4 additions. An examination of the alloy buttons revealed small glassy oxide globules on the surface of arc melted buttons but not on the EB melted buttons. The glassy substance was determined by SSMS to be a complex oxide of mostly Ca, Al, and Th. Also, the alloy buttons and ingots occasionally picked up high levels of Fe, Cu, Cr, Ni, and P during arc-melting. Because of these problems, the arc-melting method was dropped from the scale-up study.

Four heats of DOP-4 Ir-0.3% alloy, designated as DOP-4-1 to -4, were prepared by EB melting and drop-casting into $1.9 \times 1.9 \times 5.4$ cm ingots. The ingots DOP-4-1 and -2 were prepared from 100% sintered compacts, DOP-4-3 from 100% recycled heads and skulls from ingots DOP-4-1 and -2, DOP-4-4 from 50% sintered compacts and 50% recycled skull and heads from ingot DOP-4-3. The four ingots were clad in a molybdenum frame and cover plate, and hot-rolled to 0.81-mm-thick sheets between 850 and 1200°C according to the same procedure developed¹⁰ previously for fabrication of undoped Ir-0.3% W sheets used for space flight system hardware. The fabricated sheets were then sectioned — a part was used for evaluation of metallurgical and mechanical properties at ORNL; the other parts were further fabricated to 6.03-cm-diam by 0.0645-cm-thick blanks for hemishell forming at Mound Research Labs (MRL) and for capsule impact test at GE.

¹⁰D. N. Braski and A. C. Schaffhauser, *Production of Ir-0.3% W Disks and Foil*, ORNL-TM-4865 (April 1975).

The microstructure of the DOP-4 sheets was examined metallographically after short-term heat treatments between 1200 and 1800°C. A comparison of the microstructure of DOP-4 with undoped (WG) specimens heat treated at 1400 and 1500°C (Fig. 4) shows that after a 1 hr anneal at 1400 and 1500°C, the DOP-4 specimens [Fig. 4(b,d)] had an elongated grain structure with some second-phase particles distributed in the grains or on grain boundaries. By comparison, WG specimens [Fig. 4(a,c, and e)] exhibited, more or less, an equiaxed grain structure with virtually no precipitate. The degree of grain elongation in DOP-4 alloy decreased with increasing annealing time [Fig. 4(f)] and temperature. The grain size of DOP-4 and WG was measured as a function of heat treatment (see Table 9, p. 25). The DOP-4 alloy always had a finer grain structure than the undoped alloy, indicating that the DOP-4 dopants retarded grain growth in Ir-0.3% W. There is no major difference in grain structure among the four DOP-4 sheets.

The tensile properties of the DOP-4 sheets DOP-4-1 and -3 were measured at a strain rate of 5.1 mm/min ($\sim 4 \times 10^{-5}$ m/s) in vacuum at room and elevated temperatures. The tensile data (Table 6) are compared with those of undoped Ir-0.3% W alloy WG prepared from the same iridium powder and using the same melting and fabrication schedules. The strength and elongation of the DOP-4 sheets are higher than those of the undoped WG, particularly at lower temperatures. The fracture surface of the DOP-4 specimens was examined and compared with WG specimens (Table 6 and Fig. 5). At 760°C, DOP-4-1 showed TF with a high degree of secondary cracking while WG exhibited a mixed fracture mode with GBS as a major fraction. The DOP-4 material is more resistant to the intergranular fracture, confirming the results obtained from the primary study (Table 5). However, due possibly to a loss of a part of low-melting dopants (SSMS analysis, Table 2) on EB melting, the improvement of fracture behavior at 650°C is less than that of the DOP-4 alloy prepared from arc-melting.

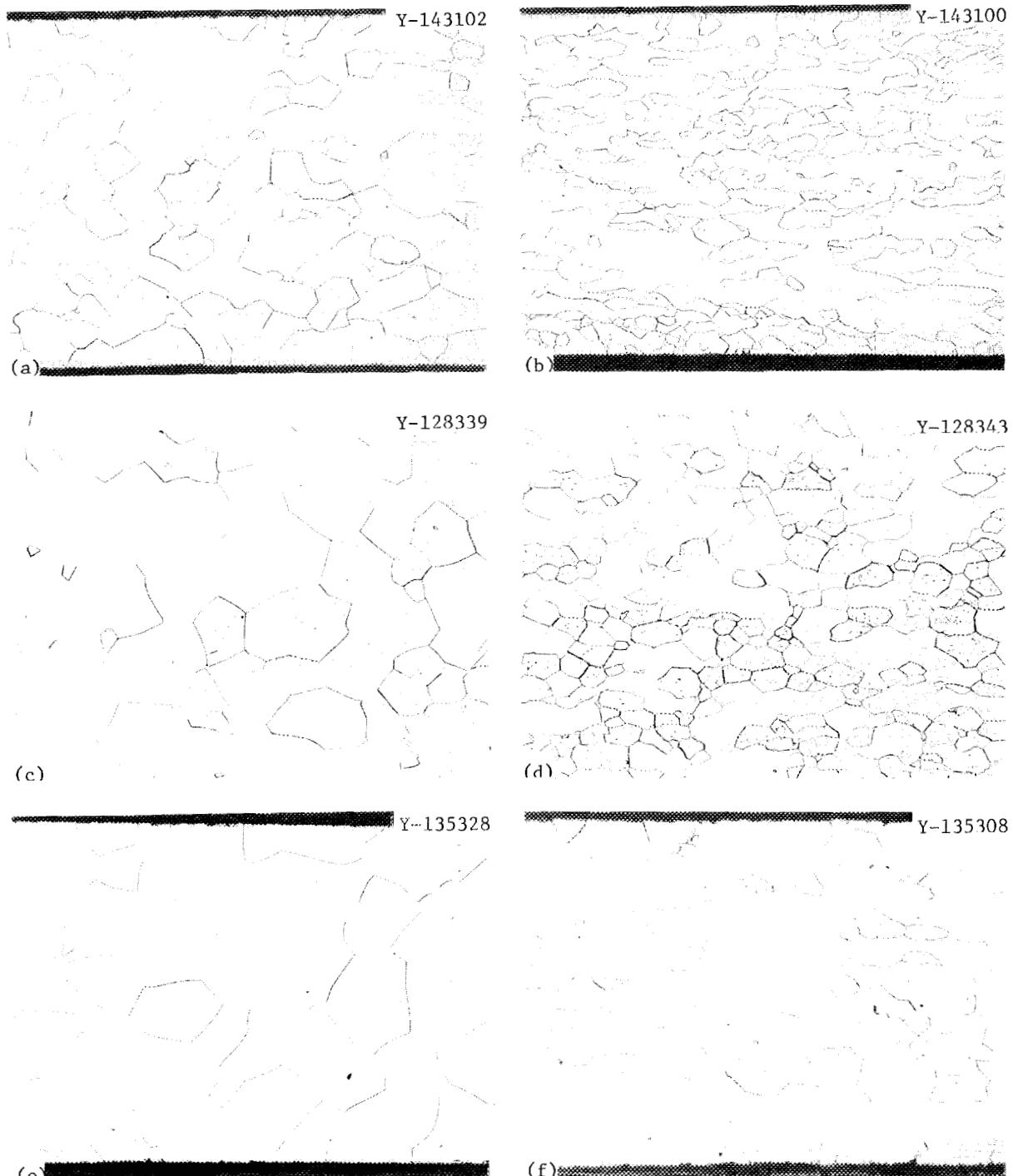


Fig. 4. Microstructures of Ir-0.3% W Alloys, 100 \times . (a) WG and (b) DOP-4 specimens annealed 1 hr at 1400°C, (c) WG and (d) DOP-4 specimens annealed 1 hr at 1500°C, (e) WG and (f) DOP-4 specimens annealed 19 hr at 1500°C. Reduced 14%.

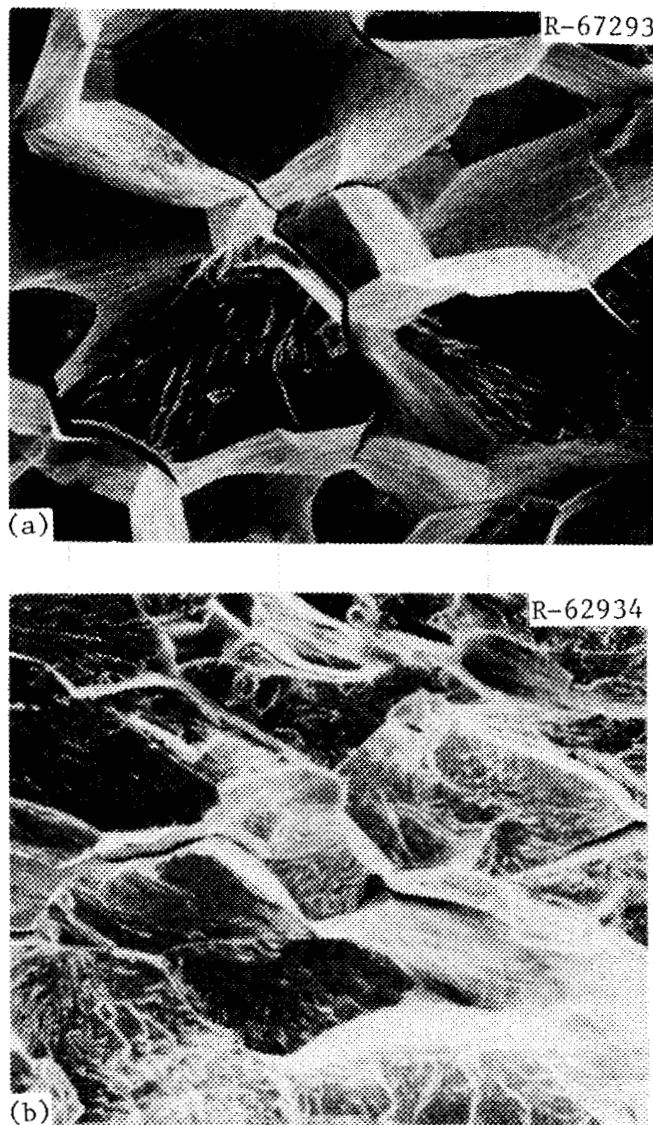


Fig. 5. Scanning Electron Microscope Fractograph of (a) Undoped WG (b) DOP-4 Tensile Specimens Fractured at 760°C, 300×.

CHARACTERIZATION OF IMPACT PROPERTIES OF DOP-4 AND UNDOPED (WG) ALLOYS

The containment materials for isotopic fuels must survive re-entry impact in case of an aborted space mission. To characterize these materials at high strain rates, equipment has been developed for impact testing at velocities to 300 fps and temperatures to 1400°C. The impact properties of Ir-0.3% W, both DOP-4 and undoped WG, were determined as a function of impact velocity, test temperature, grain size, and long-term heat treatment.

Impact Equipment

The overall view of the equipment used for the uniaxial tensile impact test is shown in Fig. 6. The impact load is imparted to the specimen by an air-driven projectile. The impact equipment consists of four major parts: (1) a 7.62-cm-diam gas gun to accelerate the projectile and its carrier at velocities to 300 fps, (2) impact chamber to test specimens in controlled atmosphere (such as argon or helium), (3) an induction heating system to heat specimens up to 1400°C, and (4) a specimen rig to load specimens for impact test. The gas gun had been calibrated by various timing systems in the Isotopes Division of ORNL and the Process Equipment Company.¹¹ A correlation¹² exists between firing pressure and the kinetic energy of projectile and its carrier (Fig. 7), where

$$\begin{aligned} p &= \text{firing pressure, psi;} \\ M &= \text{weight of projectile and its carrier, lb; and} \\ V &= \text{velocity of projectile and its carrier, fps.} \end{aligned}$$

Thus, the velocity of projectile can be determined from the firing pressure and the weight of the projectile and its carrier.

The specimens for the tensile impact test were loaded between two molybdenum alloy (TZM) pull rods and an end plate (Fig. 8). The

¹¹Tipp City, Ohio.

¹²Information and data obtained from D. W. Ramey, Isotopes Division, Oak Ridge National Laboratory.

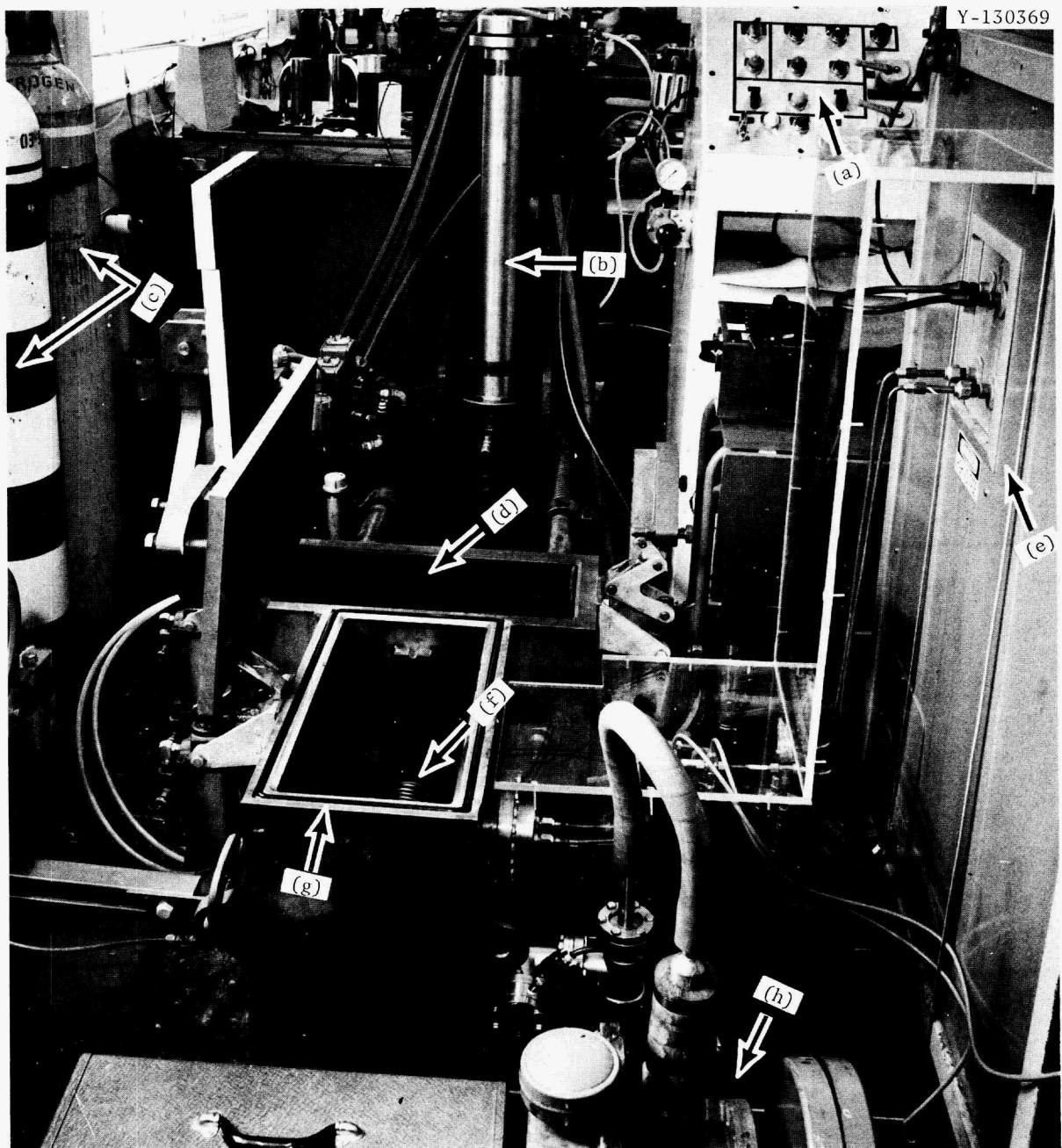


Fig. 6. High-Temperature Impact Test Equipment Used at ORNL. (a) Impact control console, (b) gas gun (in loading position) (c) gas tanks, (d) missile chamber, (e) induction unit, (f) induction coil, (g) impact chamber, and (h) mechanical pump.

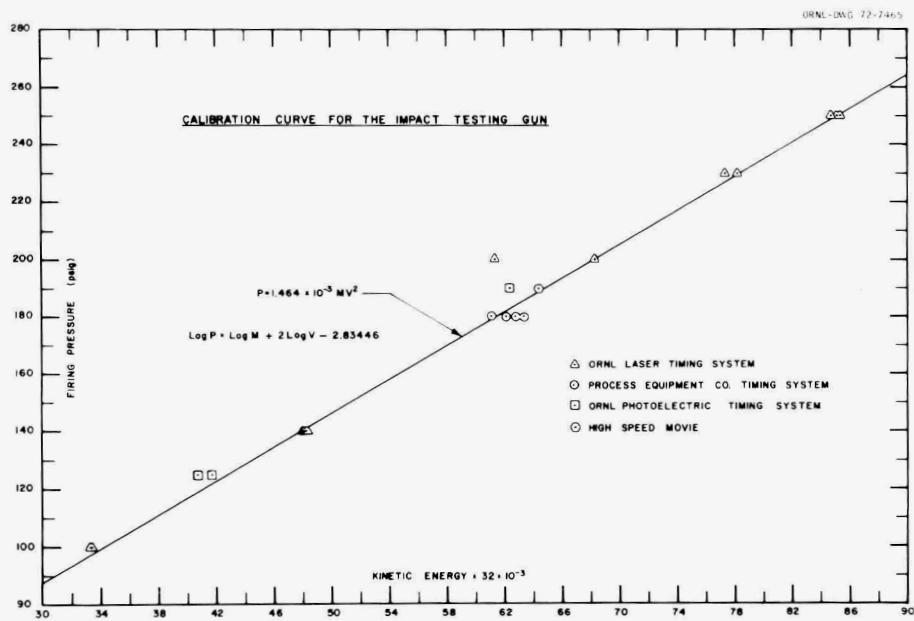


Fig. 7. Correlation Between Firing Pressure and Kinetic Energy of projectile and its carrier.

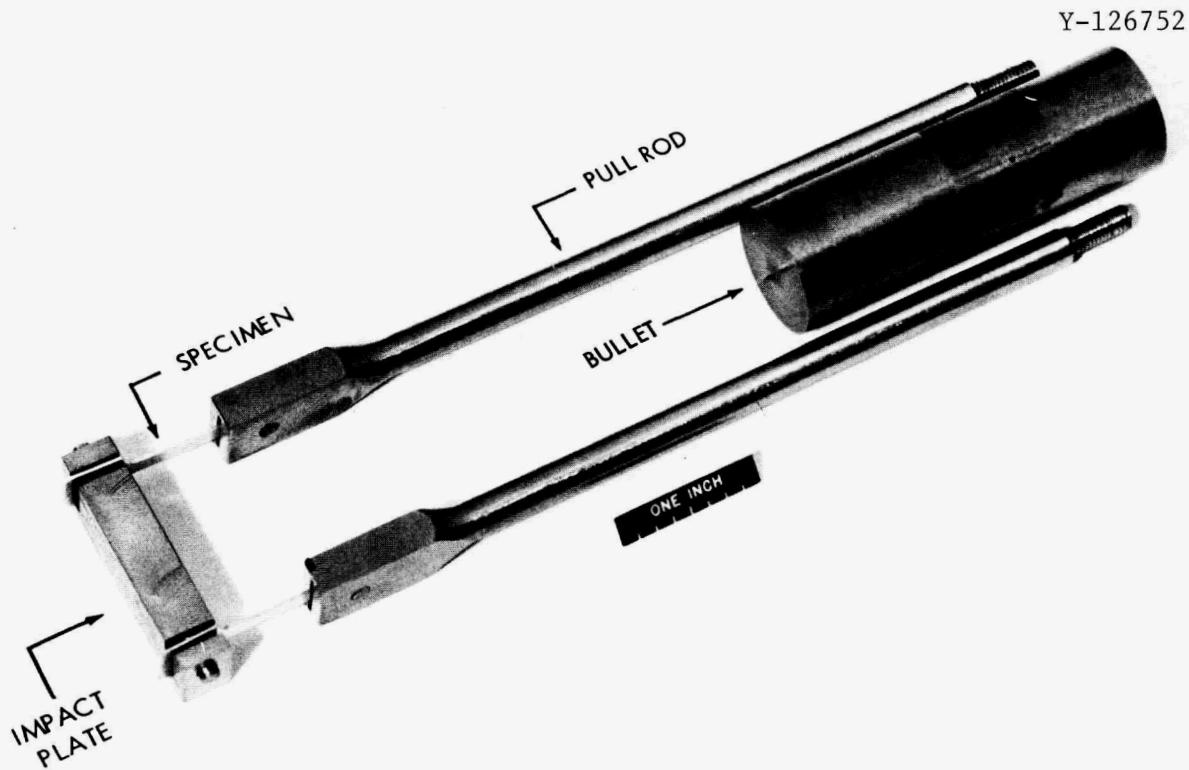


Fig. 8. Arrangement of Sheet Specimens for Tensile Impact Test.

specimens were heated by radiation from an inductively heated tantalum susceptor, and the temperature was monitored by a pt vs Pt-10% RH thermocouple centrally located on the specimen. When the specimens were heated to the desired temperature, the air-driven projectile was fired at a known velocity to break the specimens. The impact elongation and fracture mode were determined from an examination of the broken specimens.

Impact Results

The impact properties of DOP-4 and undoped (WG) Ir-0.3% W alloys were evaluated as a function of impact velocity, test temperature, grain size and heat treatment.

Impact Velocity Effect

DOP-4 and WG specimens were tested at various velocities to determine the effects of strain rate on their tensile impact properties. All the specimens were recrystallized for 1 hr at 1500°C, then impact tested at 1350°C (Table 7 and Fig. 9). Both alloys had about 45% uniform elongation and close to 100% reduction of area at a slow strain rate of 10^{-4} m/s (3.3×10^{-4} fps). The tensile impact properties of undoped WG specimens depend strongly on test velocity (Fig. 9). The ductility of WG specimens decreases sharply with an increase in test velocity, and drops to around 10% elongation at about 90 m/s (300 fps). By comparison, the impact properties of DOP-4 depend much less on velocity. DOP-4 specimens remained ductile and had 37% elongation and 90% reduction of area at 85 m/s (280 fps).

The effect of test velocity on fracture behavior of Ir-0.3% W alloys is shown in Fig. 10 and Table 7. WG specimens exhibited ductile rupture with necking to a knife edge at the slow strain rate, but fractured by TF without apparent necking at 30 m/s (100 fps). The tendency for intergranular fracture increased with further increases in test velocity, and WG specimens fractured mainly by GBS at 85 m/s (280 fps) (Fig. 10). The fracture morphology and low ductility of

Table 7. Effect of Test Velocity on Tensile Impact Properties of DOP-4 and Undoped (WG) Ir-0.3% W Sheet Specimens^a
Impact Tested at 1350°C.

Test Velocity [m/s (fps)]	Elongation (%)	Reduction of Area (%)	Fracture Mode ^b
<u>WG Specimens</u>			
10^{-4} (3×10^{-4})	43.5	~100	DR
30 (100)	34.3	40.2	Mainly TF
46 (150)	27.8	36.7	TF(Ma) and GBS(Mi)
61 (200)	16.8	24.0	Mainly GBS
85 (280)	12.6	28.0	Mainly GBS
<u>DOP-4 Specimens</u>			
10^{-4} (3×10^{-4})	46.0	~100	DR
30 (100)	41.5	75.0	DR, TF
46 (150)	43.1	94.0	DR
61 (200)	36.3	95.0	DR
85 (280)	37.6	94.0	DR

^aSpecimens annealed 1 hr at 1500°C before impact test.

^bGBS = grain-boundary separation, TF = transgranular fracture, DR = ductile rupture, Ma = major fraction, and Mi = minor fraction.

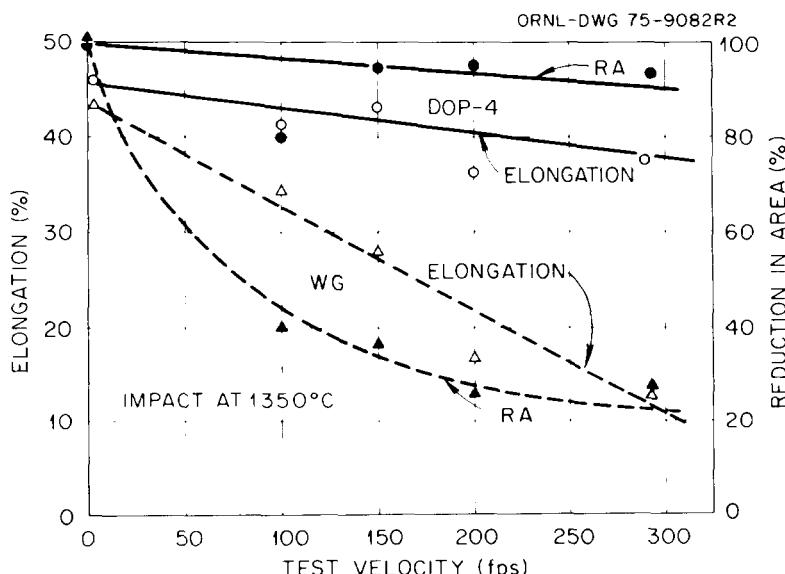


Fig. 9. Impact Elongation and Reduction in Area of DOP-4 and WG Specimens as a Function of Test Velocity.

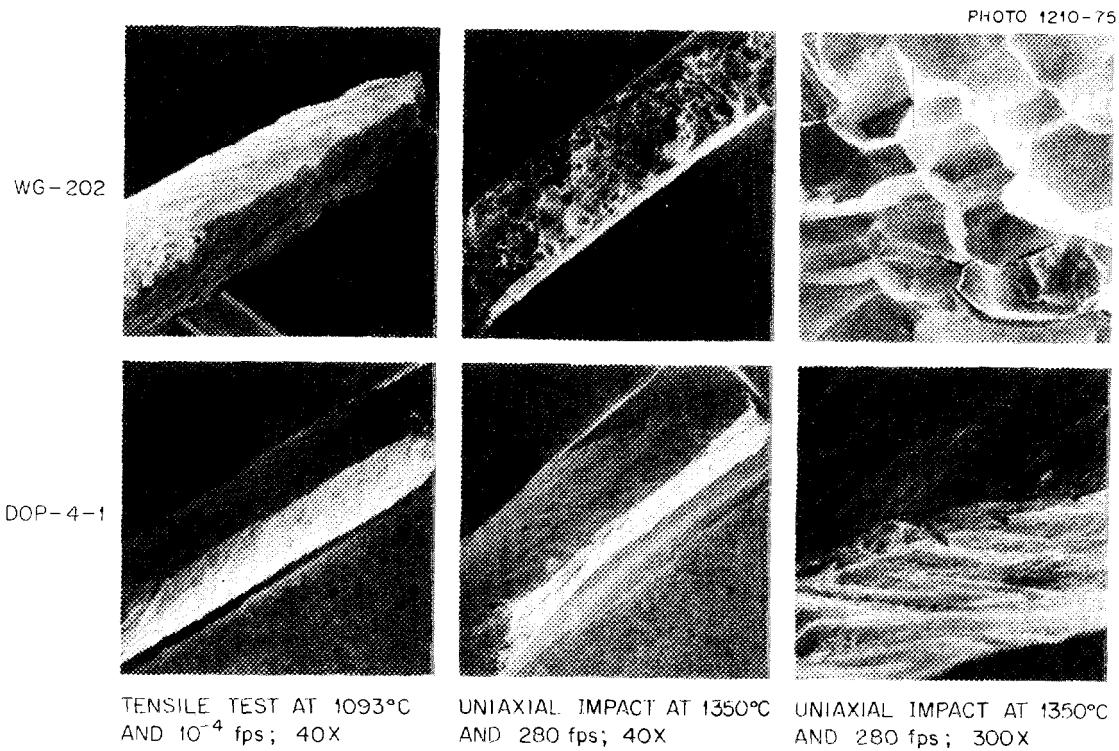


Fig. 10. Effect of Strain Rate on Fracture Mode of Undoped (WG-202) and Doped (DOP-4-1) Ir-0.3% W.

WG specimens tensile impact tested at 1350°C and 85 m/s are similar to those observed on the Ir-0.3% W PICS of FSA impact tested under the same conditions; therefore, there is no doubt that the brittle fracture associated with the impact of PICS is due to the high strain rate effect. DOP-4 specimens showed ductile rupture with a high reduction of area at all strain rates (Fig. 10). Thus, the DOP-4 alloy has much better impact resistance.

Impact Temperature Effects

WG and DOP-4 specimens were heat treated 1 hr at 1500°C, and impact tested in the range of 950-1350°C to determine the effects of temperature on impact properties. All the specimens were impacted at 85 m/s (280 fps) (Table 8). Impact elongation is plotted as a function of test temperature (Fig. 11). The impact properties of WG and DOP-4 specimens are not sensitive to temperature above 1200°C; however, below this temperature

Table 8. Effect of Test Temperature on Tensile Impact Properties of DOP-4 and Undoped (WG) Ir-0.3% W Sheet Specimens^a
Tested at 85 m/s (280 fps)

Temperature (°C)	Elongation (%)	Reduction of Area (%)	Fracture Mode ^b
<u>WG Specimens</u>			
1350	12.6	28	Mainly GBS
1250	12.5	22	Mainly GBS
1250	13.7	28	Mainly GBS
1250	14.3	31	Mainly GBS
1150	8.6	18	Mainly GBS
1050	5.4	13	Mainly GBS
950	4.5	12	Mainly GBS
<u>DOP-4 Specimens</u>			
1350	37.6	94	DR
1250	38.2	82	DR
1150	21.5	54	TF
1050	15.2	44	TF
1050	13.6	25	GBS and TF
950	10.3	24	Mainly GBS

^aSpecimens heat-treated 1 hr at 1500°C before impact test.

^bGBS = grain-boundary separation, TF = transgranular fracture, and DR = ductile rupture.

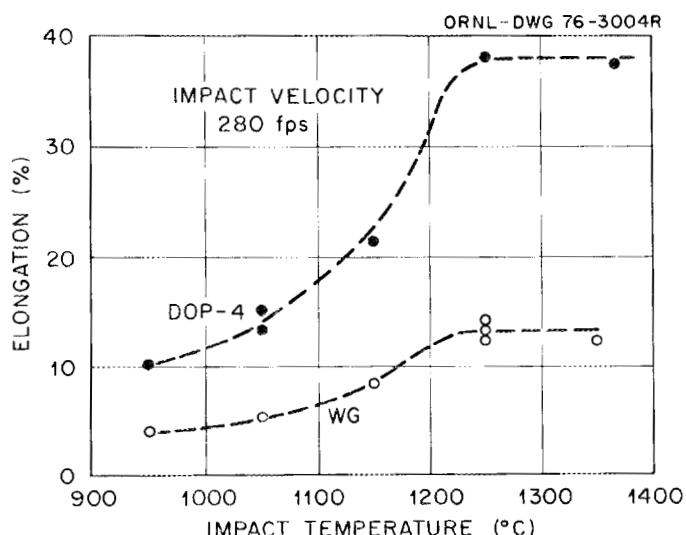


Fig. 11. Impact Elongation as a Function of Impact Temperature for DOP-4 and Undoped (WG) Ir-0.3% W Alloys.

the ductility decreases for both alloys. DOP-4 was much more ductile than WG at all temperatures. The WG specimens fractured mainly by GBS at these temperatures, while the fracture mode of DOP-4 changed from ductile rupture above 1200°C to mixed GBS and TF at 1050°C and to mainly GBS at 950°C.

Grain Size Effect

The impact properties of DOP-4 and WG may be given as a function of grain size produced by short-term heat treatments between 1300–1800°C (Table 9 and Fig. 12). All the specimens were impact tested under the same conditions, for example, 85 m/s (280 fps) and 1350°C. The impact elongation of WG specimens decreases continuously with increasing grain size, and reduces to about 4% with 2 or 3 grains across the specimen. The ductility of DOP-4 is not sensitive to grain size in the fine grain size range, but the ductility decreases in the coarse grain size range. At a given grain size, the impact ductility of DOP-4 is much better than for the WG specimens. This strongly suggests the possible improvement of grain boundary strength in DOP-4 alloy through segregation of dopants at the boundaries.

Table 9. Effect of Grain Size on Tensile Impact Properties of DOP-4 and Undoped (WG) Ir-0.3% W Sheet Specimens Tested at 85 m/s (280 fps) and 1350°C

Temperature (°C)	Time (hr)	Heat Treatment		Elongation (%)	Reduction of Area (%)	Fracture Mode ^b
		Average	Range			
WG Specimen						
1300	1	19.7	21-17	18.7 ^c	27 ^c	GBS and TF
1400	1	13.8	16-12	14.5	27	GBS (Ma) and TF (Mi)
1500	1	10.7	12-7	12.6	28	Mainly GBS
1500	19	5.8	8-3	10.9	23	Mainly GBS
1500 ^d	19	5.8	8-3	10.5	24	Mainly GBS
1700 ^d	1	4.1	6-2	8.4	19	Mainly GBS
1800 ^d	1	2.4	4-1	2.2	7	Completely GBS
1800 ^d	1	2.4	4-1	4.6	5	Completely GBS
DOP-4 Specimens						
1300	1	26.9	30-25	36.0	94	DR
1300	1	26.9	30-25	34.1 ^c	74 ^c	DR
1400	1	25.0	27-23	36.2	90	DR
1500	1	19.2	22-16	37.6	94	DR
1500 ^d	19	9.9	12-9	26.8	60	TF, DR
1700 ^d	1	6.1	9-4	16.5	46	TF (Ma) and GBS (Mi)
1800 ^d	1	3.8	5-2	11.0	19	Mainly GBS

^aNumber of grains across a 0.64-mm-thick (0.025-in.) sheet.

^bCAB = grain-boundary separation; TF = transgranular fracture; Ma = major fraction; Mi = minor fraction; and DR = ductile rupture.

^cImpact tested at 280 fps at 1250°C.

^dThe specimen was subjected to a 1-hr heat treatment at 1500°C before and after the high-temperature anneal at 1700 or 1800°C.

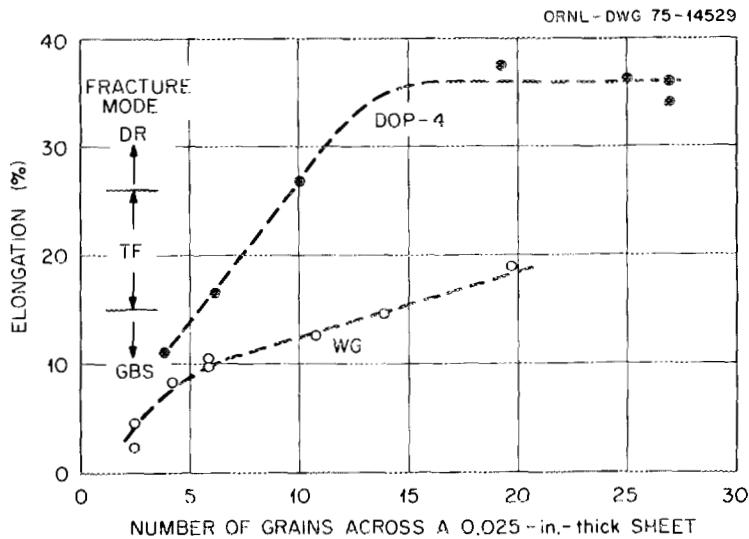


Fig. 12. Impact Elongation and Fracture Mode as a Function of Grain Size for DOP-4 and Undoped (WG) Ir-0.3% W Alloys.

Both DOP-4 and WG alloys may lose their ductility when the grain size is sufficiently large since both ductility curves may extend to the origin (Fig. 12). This seems to agree well with the biaxial-impact data obtained¹³ recently at LASL. Thus, it is the grain size, rather than grain-boundary chemistry that controls the impact properties of Ir-0.3% W alloys in the coarse grain size range. Since the dopant additions also retard grain growth, the combined effects result in the DOP-4 alloy having an impact ductility a factor of 2 to 3 higher than for the undoped alloy for any annealing condition.

The specimens with more than 25% elongation generally fractured by ductile rupture, while those specimens with less than 15% elongation fractured by GBS. Between these limits, the fracture mode was predominantly by TF.

Long-Term Anneal Effect

DOP-4 and WG specimens were annealed to 2000 hr to 1330°C, the MHW fuel-clad temperature, and then impact tested at 1350°C and 85 m/s (280 fps) (Table 10 and Fig. 13).

¹³Private communications with Sig. Hecker, LASL, 1975-1976.

Table 10. Effect of Long-Term Anneal at 1330°C on the Impact Properties of DOP-4 and Undoped (WG) Ir-0.3% W Sheet Specimens Impacted at 85 m/s (280 fps) and 1350°C

Annealing Time (hr)	Specimen	Grain Size ^a	Impact Elongation (%)	Reduction of Area (%)	Fracture Mode ^b
1 ^c	WG	19.7	18.7	27	GBS and TF
1 ^c	DOP-4	26.9	36.0	94	DR
200	WG	9.0	13.7	28	Mainly GBS
200	DOP-4				
480	WG	8.3	8.4	24	Mainly GBS
480	DOP-4	15.2	26.5	9.0	DR
980	WG	7.3	8.6	20	Mainly GBS
980	DOP-4	11.0	27.2	88	DR
2000	WG	4.7	7.8	19	Mainly GBS
2000	DOP-4	8.5	24.0	61	Mainly TF

^aNumber of grains across a 0.64-mm-thick (0.025-in.) sheet.

^bGBS = grain boundary separation, TF = transgranular fracture, and DR = ductile rupture.

^cAnneal 1 hr at 1300°C.

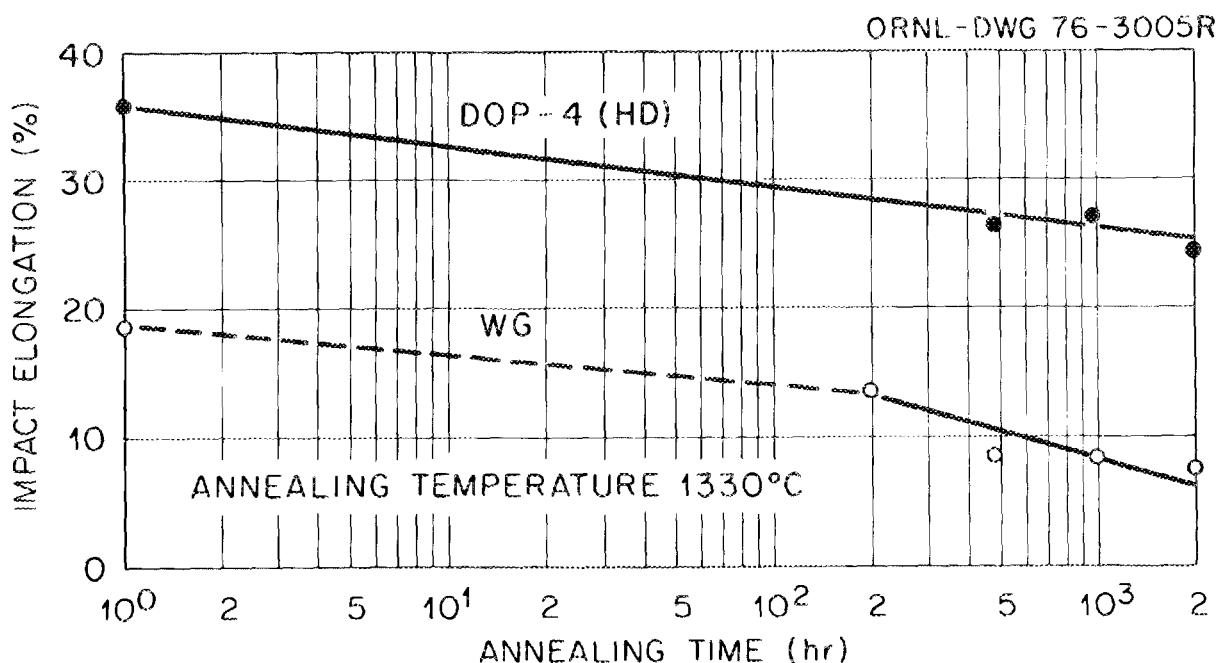


Fig. 13. Effect of Long-Term Annealing at 1330°C on Impact Elongation of WG and DOP-4, Ir-0.3% W Alloys.

The grain size of the specimens was also measured (Table 10). The impact elongation of both alloys decreases with annealing time, and this decrease is apparently due to grain growth. The WG material had a marginal ductility of 15% elongation after being annealed for only 30 hr at 1300°C, while DOP-4 material remains ductile after annealing for 2000 hr. Thus, DOP-4 material is much more resistant to degradation due to long-term heat treatment at 1330°C. A detailed study of grain-growth behavior in doped and undoped Ir-0.3% W alloys has been reported.¹⁴

Results (Tables 7-10 and Figs. 9-13) indicate the impact properties of DOP-4 Ir-0.3% W alloy are far superior to those of the undoped alloy.

Correlation of Uniaxial Tensile Impact Test with MHW/FSA Impact Test

Postimpact containment shells (PICS) were made from DOP-4 material to confirm the superior impact properties in actual hardware configurations. Blanks from DOP-4-1 to -4 sheets were fabricated into hemishells and assembled with a 76% dense ThO₂ (fuel simulant) and then encased in a 10-mm-thick GIS at Monsanto Research Corporation (MRC) and GE. The hemishells were heat treated for 16.5 hr at 1500°C before welding into the postimpact sphere assembly. The three FSA, designated as MHT-90, -91, and -92, were impact tested against a granite target at 1370°C at a velocity of 90-95 m/s (295-304 fps). Table 11 lists the impact parameters and the postimpact results obtained at GE, and the grain-size measurement at ORNL. A detailed metallographic examination of these three impact tested FSA has been reported recently.¹⁵

All three DOP-4 PICS survived the MHW/FSA impact with no indication of breaching or cracking. Figure 14(b) shows the impact face of MHT-91. The normal deformation at the region of maximum hoop strain ranged between 9.3 and 11.2%. Figure 14(a) shows the impact face of a WG

¹⁴D. E. Harasyn and A. C. Schaffhouser, *Grain Growth in Ir-0.3% W Alloys*, ORNL-5233 (January 1977).

¹⁵E. M. Cramer and S. S. Heckler, *Postimpact Examinations of the DOP-4 Ir Shells from Simulant Fuel Sphere Assemblies*, LA-6176-MS, LASL, (February 1976).

Table 11. FSA Impact Data^a for DOP-4 Ir-0.3% W Alloy

	MHT-90	MHT-91	MHT-92
<u>PICS Identity</u>			
Female	DOP-4-3	DOP-4-2	DOP-4-4
Male	DOP-4-1	DOP-4-2	DOP-4-4
<u>Impact Condition</u>			
Temperature, °C	1370	1370	1370
Rotation, degrees from normal hit	50	4	48
Velocity, m/s	95	90	89.6
Heat Treatment at 1500°C, hr	16.5	16.5	16.5
<u>Post Impact Results</u>			
Average diameter, mm	44.50	44.68	45.34
Average height, mm		28.75	28.58
Average deformation, mm	9.3	9.6	11.2
Dye penetration examination	negative ^b	negative	negative
30X examination	negative ^b	negative	negative
Grain Size Measurement ^c	11.4	10.8	11.1

^aData obtained from General Electric Company, except for the grain size measurement.

^bMicrocracking in regions having dye penetrant indications before impact; no other cracking at 30X or by dye penetrant.

^cNumber of grains per thickness; measured at ORNL.

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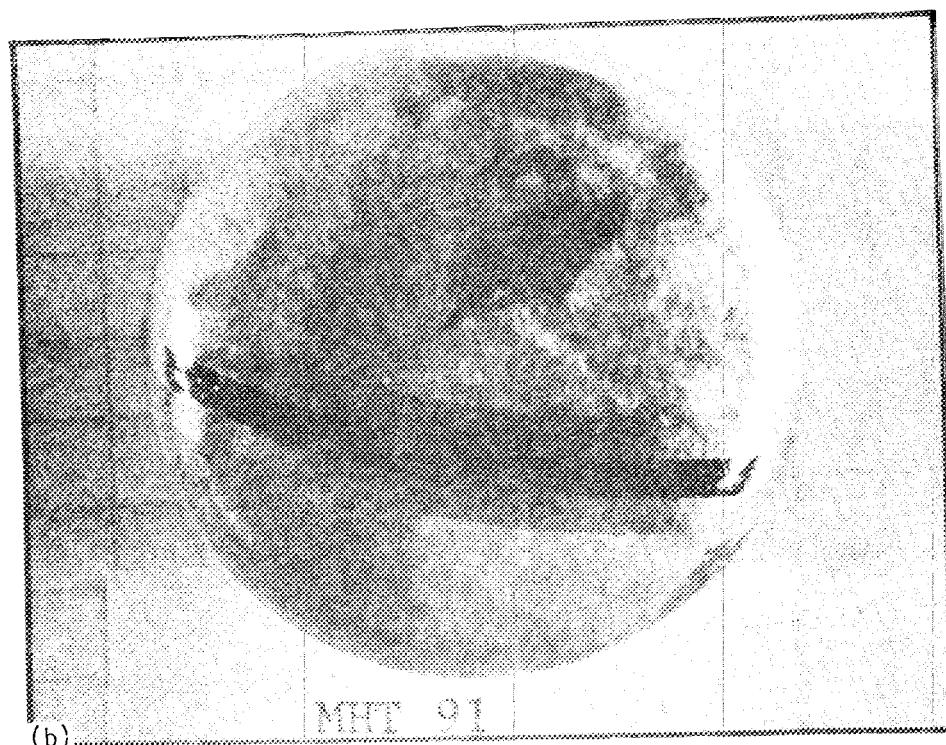
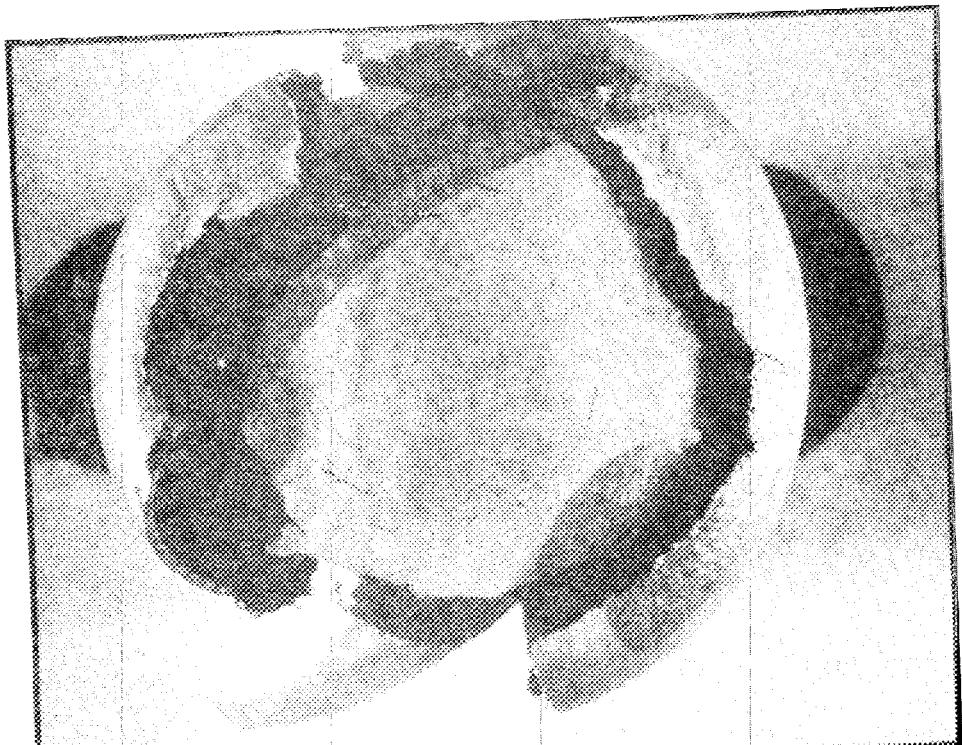


Fig. 14. Impact Face of FSA/MMT Impact Tested at 1370°C and 90 m/s.
(a) Undoped WG material (MHT-94); (b) DOP-4 material (MHT-91).

Ir-0.3% W FSA under similar conditions. Fingerprint cracks and hoop-strain failure seen after tests of some undoped Ir-0.3% W FSA were not evident in these DOP-4 tests. A total of 4 WG FSA were impact tested; all of which failed with different degrees of breaching. This comparision confirms our uniaxial tensile impact data that the impact performance of DOP-4 material is far superior to that of the undoped material. Therefore, the development of the DOP-4 alloy greatly increases the safety margin of the MHW heat sources whose next mission is a space probe to Jupiter and Saturn in 1977.

The MHW/FSA impact test is expensive and time-consuming. In addition, it is a multicomponent test and variations in fuel and GIS properties or other interactions may influence the behavior of this iridium. The uniaxial impact test would serve as a quick and useful method to predict the outcome of the MHW/FSA impact test if a suitable correlation could be established between the two tests.

An effort was made to correlate the impact ductilities with grain size of materials from the tests. Before the FSA impact test, the Ir-0.3% W PICS were heat treated 16 to 19 hr at 1500°C. This heat treatment is necessary to simulate the grain growth in a period of ground operation. This heat treatment produced about 9-13 grains across the PICS thickness [Figs. 15 and 4(f)]. The DOP-4 specimen with that grain size had more than 25% impact elongation (Fig. 12) and fractured by ductile rupture when impact tested at 1350°C and 85 m/s. The success in the DOP-4 sphere impact tests at GE suggests that 25% impact elongation is sufficient for cladding materials to survive the FSA/MHW impact. The FSA heat treatment produced 5-7 grains across the PICS of WG material [Fig. 4(e)]. The impact test results (Fig. 12) indicate that the undoped Ir-0.3% W with that grain size had about 9-12% elongation and its fracture mode was mainly GBS. The failures in the WG FSA impact tests suggest that more than 12% impact ductility is needed for surviving these tests. On the FSA impact test, the nominal deformation at the hoop strain region usually ranges between 8-12% but sometimes may reach as high as 14-15%. Thus, this comparison, indicates that cladding materials should have 15% or more ductility in order to ensure success in the MHW/FSA impact test.

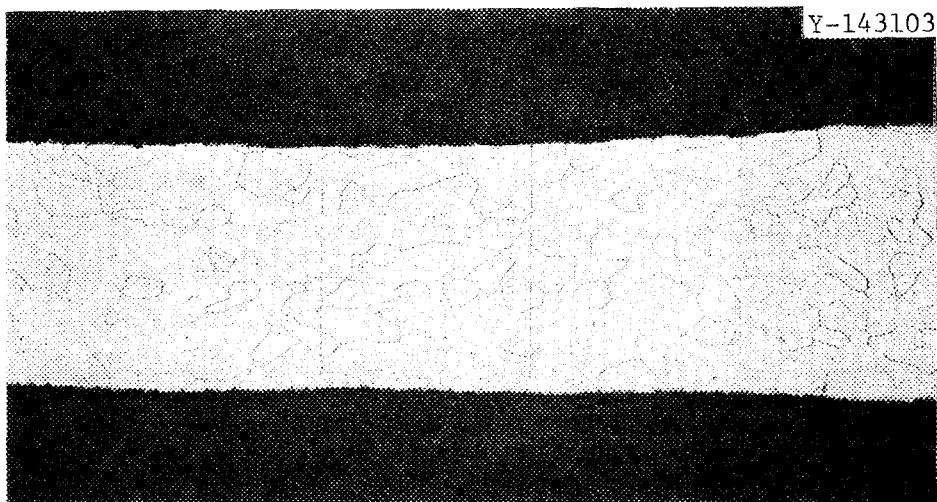


Fig. 15. Microstructure of DOP-4 Ir-0.3% W PICS Sectioned from MHT-92. 50 \times .

GENERAL DISCUSSION

Function of DOP-4 Dopants

The results presented in the previous sections indicate that the total of 240 ppm DOP-4 dopants (30 ppm Th, 40 Al, 80 Fe, 16 Ni, and 75 Rh) added to the DOP-4 alloy suppresses the brittle grain-boundary fracture and greatly improves the impact properties of Ir-0.3% W. In addition, the DOP-4 dopants raise the recrystallization temperature of Ir-0.3% W and retard its grain growth at high temperatures. Evidence obtained so far suggests that the DOP-4 dopants affect the metallurgical and mechanical properties of Ir-0.3% W through segregation of the dopants to grain boundaries, and precipitation of second-phase particles.

Segregation of the Dopants to Grain Boundaries

Chemical analyses of the intergranularly-fractured surfaces by SSMS have indicated¹⁶ that thorium tends to segregate strongly near the grain boundaries of iridium and Ir-0.3% W alloys. The thorium

¹⁶C. T. Liu and H. Inouye, *Study of Iridium and Iridium-Tungsten Alloys for Space Radioisotopic Heat Sources*, ORNL-5240 (December 1976).

concentration in the vicinity of the boundaries is higher than its bulk value by four orders of magnitude. SSMS analysis has also shown that aluminum iron tend to segregate near the grain boundaries, but not as strongly as thorium. The segregation of thorium to the grain boundaries has been confirmed by auger electron spectroscopy (AES). White and Clausing¹⁷ at ORNL have found that the thorium concentration at the grain boundary of the DOP-4 material is of the order of 3-5 at. %, as compared with only 30 ppm (by wt) in the bulk. Inert-ion sputtering at the fracture surface indicates that the thorium is concentrated only within a few atom layers (<10 atm layers) of the grain boundary. Similar AES studies at LASL have shown¹⁸ about the same amount of thorium segregation at the grain boundary of the DOP-4 alloy. No thorium is detected at the grain boundary of undoped (WG) specimens.

The exact role played by dopants at the grain boundary is not yet clear; however, their presence improves the coherent strength of the boundary. As a consequence, the DOP-4 material is more resistant to grain-boundary fracture and has superior impact properties. The impact properties of the DOP-4 and undoped alloys will be discussed in detail later.

Precipitation of Second-Phase Particles

Small amounts of second-phase particles appear in the DOP-4 specimens but not in the undoped specimens (Fig. 4). Since the solubility of Fe, Ni and Rh in iridium is quite high, the second-phase particles are most probably an iridium compound containing thorium and possibly aluminum. The published iridium-thorium phase diagram indicates¹⁹ only very limited solubility of thorium in iridium and the first compound formed at the iridium-rich end is ThIr₅. Numerous precipitates probably ThIr₅, are clearly visible²⁰ in the Ir-0.3% W alloys

¹⁷Private communication with C. L. White and R. E. Clausing, 1975.

¹⁸R. D. Baker, Los Alamos Scientific Laboratory., private communications during 1974-1976.

¹⁹F. A. Shunk, "Ir-Ti He Iridium-Titanium" p. 466, *Constitution of Binary Alloys, Second Supplement*, McGraw-Hill Book Company, N.Y., 1969.

²⁰C. T. Liu and H. Inouye, unpublished results.

containing high levels of thorium. The particles in DOP-4 seem to be stable and do not grow appreciably on heat treatment at 1500°C (Fig. 4).

The DOP-4 specimens distinctly show an elongated grain structure on short-term heat treatments at 1500°C or below (Fig. 4). The elongated structure probably results from the restriction of grain growth by the particles which line up along the fibrous direction during hot fabrication. The grains in the DOP-4 specimens become equiaxed after being annealed 19 hr at 1500°C [Fig. 4(f)]; however, the grain size is still much finer than that of the undoped alloy. Probably the slower grain growth in the DOP-4 alloy is due to both particle and impurity atom pinning at the grain boundary. As noted previously, a small grain size is one of the factors that contributes to the better impact properties of the DOP-4 alloy.

Effects of Grain Size, Dopant Addition, Impact Velocity and Test Temperature on Impact Properties

The tensile and impact data in the previous sections indicate that the deformation and fracture behavior of Ir-0.3% W is sensitive to grain size, dopants, impact velocity and test temperature. The brittle fracture associated with GBS in these materials is promoted with decreasing test temperature and increasing strain rate or grain size. All these results can be correlated on the basis of the stress concentration on grain boundaries through a model of pile up of glide dislocations. The stress concentration due to pile up of dislocations²¹ is

$$[\sigma - \sigma_0][d/s]^{1/2} ,$$

where σ is the applied stress, σ_0 the frictional stress against the motion of dislocations on the slip plane, d the grain diameter, and s a distance beyond the tip of a pile up.

²¹J. D. Eshelby, F. C. Frank, and F.R.N. Nabarro, "Equilibrium of Linear Arrays of Dislocations," *Philos. Mag.* 42: 351 (1951).

Intergranular fracture will occur if the stress concentration at a grain boundary reaches the coherent strength of the boundary σ_c :

$$(\sigma_f - \sigma_0) (d/s)^{1/2} = \sigma_c , \quad (1)$$

where σ_f is the fracture stress. To correlate the stress with strain, ϵ , the Ludwik-type equation²² is used:

$$\sigma = \sigma_i + K\epsilon^n , \quad (2)$$

where σ_i , K and n are material constants. At fracture,

$$\sigma_f = \sigma_i + K\epsilon_f^n , \quad (3)$$

where ϵ_f is the fracture strain. Combination of Eqs. (1) and (3) and rearrangement gives the final expression:

$$\epsilon_f = [(\sigma_0 - \sigma_i)/K + (\sigma_c^{S^{1/2}}/Kd^{1/2})]^{1/n} . \quad (4)$$

For simplicity, let $n = 1$ for the case of linear work-hardening, then Eq. (4) reduces to,

$$\epsilon_f = (\sigma_0 - \sigma_i)/K + (\sigma_c/K (s/d)^{1/2} . \quad (5)$$

Eq. (5) has been used to predict the impact properties of the Ir-0.3% W alloys, and the predictions are now compared with the experimental results previously presented.

Grain-Size Effect

At a given test condition, Eq. (5) predicts that ϵ_f will decrease with an increase in grain size d . This is in agreement with the data

²²P. Ludwik, *Elemente der Technologischen Meibank*, Verlag Von Julius, Springer, Berlin, 1909.

correlating impact elongation and fracture mode with grain size (Fig. 12). Furthermore, the impact elongation (ϵ_f) should be a function of square root of grain size, d , according to Eq. (5). A linear relation exists between ϵ_f and $d^{-1/2}$ for the WG specimens which showed grain-boundary fracture when impact tested at 1350°C and 280 fps (Fig. 16). The linear relation also holds quite well for the DOP-4 specimens, even though they exhibited ductile rupture in the fine-grain-size range.

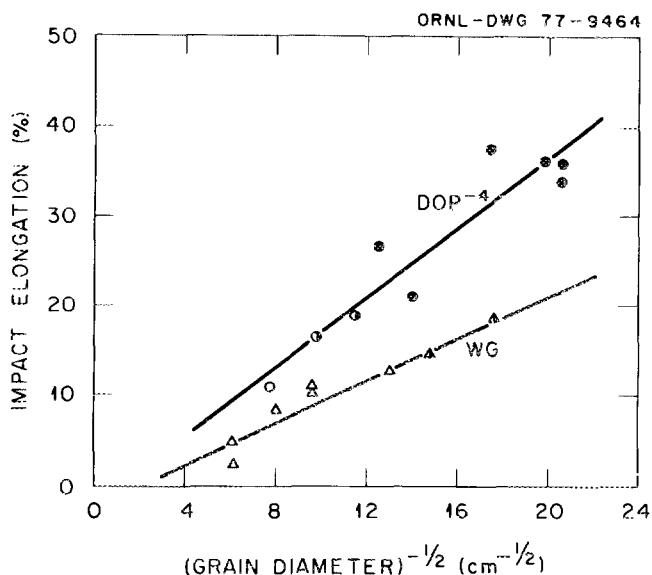


Fig. 16. Impact Elongation as a Function of Square Root of Grain Diameter for DOP-4 and Undoped (WG) Ir-0.3% W Alloys. Open symbols for mainly GBS, filled symbols for mainly TF, and half-filled symbols for mixed fracture of GBS and TF.

Dopant Effect

As already discussed in this section, the DOP-4 dopants such as thorium segregate to and improve the coherent strength of the grain boundary. This is also manifested by measuring the slope of the two curves in Fig. 16, which shows that σ_c [Eq. (5)] for the DOP-4 alloy may be higher than the undoped alloy by 60%. Also, the dopants retard grain growth in the DOP-4 alloy through dopant segregation and precipitation of second-phase particles; thus in DOP-4 alloy the grain structure

was much finer than that of the undoped alloy under the same heat treatment condition. In Eq. (5), both an increase in σ_c and a decrease in d , improve the fracture ductility. By a combination of these two effects the impact properties of DOP-4 are greatly improved.

Test Temperature and Velocity Effects

Both σ_c and K increase with a drop in test temperature or an increase in strain rate.²³ Both of these lower ϵ_f in Eq. (5). This prediction also agrees with the impact data (Figs. 9 and 11).

In summary, the impact properties of the DOP-4 and undoped Ir-0.3% W alloys can be correlated by Eq. (5), which is derived on the basis of stress concentration at grain boundaries.

SUMMARY AND CONCLUSIONS

The Ir-0.3% W alloy has been doped with minor alloying additions for the purpose of developing improved cladding material for space radioisotopic heat sources operating up to 1450°C. The dopants were thoroughly mixed with iridium powders, compacted, sintered, then arc or electron-beam melted. A total of 10 doped alloys were prepared; the DOP-4 alloy containing 40 ppm Al, 30 Th, 80 Fe, 16 Ni and 75 Rh was the most resistant to brittle fracture associated with grain-boundary separation on tensile testing at slow strain rates. In addition, the DOP-4 dopants improve the fabricability, raise the recrystallization temperature, and retard the grain growth at high temperatures.

Impact-testing equipment has been developed for testing materials in controlled atmospheres at temperatures to 1400°C at impact velocities as high as 90 m/s (300 fps) (which is equal to a strain rate of 7200/s. The impact properties of DOP-4 and undoped (WG) Ir-0.3% W alloys were determined as a function of grain size, test

²³R. W. Swindeman, *Low-Strain Tensile Behavior of Type 304 Stainless Steel (Heat 9T2796)*, ORNL/TM-5245 (February 1976).

temperature, impact velocity and long-term heat treatment. The brittle grain-boundary fracture is always promoted in these alloys by a decrease in test temperature or an increase in strain rate or grain size. All these results can be correlated by Eq. (5), which is derived on the basis of stress concentration at grain boundaries.

The impact test results (Tables 7-11 and Figs. 9-15) clearly indicate that the impact properties of the DOP-4 alloy are far superior to those of the undoped (WG) alloys. The improvement is due to combined effects of segregation of beneficial dopants (such as thorium) to grain boundaries and a refinement of grain structure through dopant segregation and precipitation of second-phase particles. The superior impact properties of DOP-4 alloy, measured by uniaxial impact testing, are confirmed by the biaxial impact tests at LASL and by the MHW/FSA impact tests at GE. The development of the DOP-4 alloy greatly increases the safety margin of the MHW heat sources; therefore, the NRA/ERDA has decided recently to use this alloy to contain the heat source for the Mariner/Jupiter-Saturn mission scheduled for 1977.

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